



# *Thermodynamic Properties*

- ◆ **Property Table --**  
**from direct measurement**
- ◆ **Equation of State --**  
**any equation that relates  
P, v, and T of a substance**



# Exercise 3-12

**A bucket containing 2 liters of R-12 is left outside in the atmosphere (0.1 MPa)**

- a) What is the R-12 temperature assuming it is in the saturated state.**
- b) the surrounding transfer heat at the rate of 1KW to the liquid. How long will take for all R-12 vaporize?**

**See R-12 (diclorindifluormethane) on Table A-2**

# Solution - page 1

## Part a)

From table A-2, at the saturation pressure of 0.1 MPa one finds:

- $T_{\text{saturation}} = -30^{\circ}\text{C}$
- $v_{\text{liq}} = 0.000672 \text{ m}^3/\text{kg}$
- $v_{\text{vap}} = 0.159375 \text{ m}^3/\text{kg}$
- $h_{\text{lv}} = 165 \text{ KJ/kg}$  (vaporization heat)

# Solution - page 2

## Part b)

The mass of R-12 is  $m = \text{Volume}/v_L$ ,  
 $m = 0.002/0.000672 = 2.98 \text{ kg}$

The vaporization energy:

Evap = vap energy \* mass =  $165 * 2.98 = 492 \text{ KJ}$

Time = Heat/Power =  $492 \text{ sec or } 8.2 \text{ min}$



***GAS PROPERTIES***



# *Ideal -Gas Equation of State*

$$PV = nR_u T; \quad n = \frac{M}{\text{mol}}$$

**Universal gas constant is given on**

$$\begin{aligned} R_u &= 8.31434 \text{ kJ/kmol-K} \\ &= 8.31434 \text{ kPa-m}^3/\text{kmol-k} \\ &= 0.0831434 \text{ bar-m}^3/\text{kmol-K} \\ &= 82.05 \text{ L-atm/kmol-K} \\ &= 1.9858 \text{ Btu/lbmol-R} \\ &= 1545.35 \text{ ft-lbf/lbmol-R} \\ &= 10.73 \text{ psia-ft}^3/\text{lbmol-R} \end{aligned}$$

# *Example*

Determine the particular gas constant for air (28.97 kg/kmol) and hydrogen (2.016 kg/kmol).

$$\mathbf{R_{air}} = \frac{\mathbf{R_u}}{\mathbf{M}} = \frac{8.1417 \frac{\mathbf{kJ}}{\mathbf{kmol-K}}}{28.97 \frac{\mathbf{kg}}{\mathbf{kmol}}} = 0.287 \frac{\mathbf{kJ}}{\mathbf{kg-K}}$$

$$\mathbf{R_{hydrogen}} = \frac{8.1417 \frac{\mathbf{kJ}}{\mathbf{kmol-K}}}{2.016 \frac{\mathbf{kg}}{\mathbf{kmol}}} = 4.124 \frac{\mathbf{kJ}}{\mathbf{kg-K}}$$





# ***Ideal Gas “Law” is a simple Equation of State***

$$\mathbf{PV = MRT}$$

$$\mathbf{Pv = RT}$$

$$\mathbf{PV = NR_u T}$$

$$\frac{\mathbf{P_1 V_1}}{\mathbf{T_1}} = \frac{\mathbf{P_2 V_2}}{\mathbf{T_2}}$$



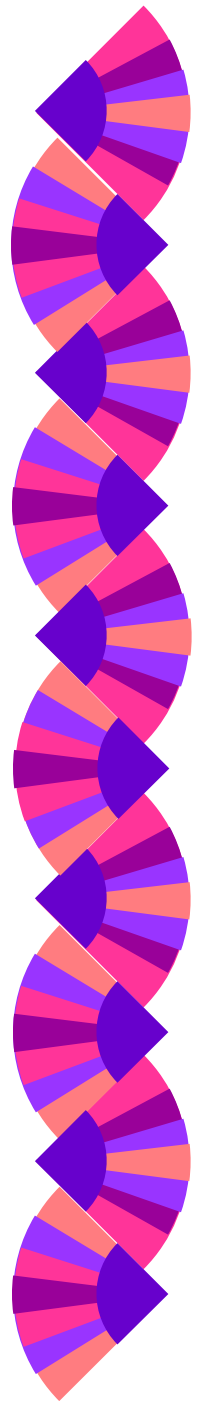
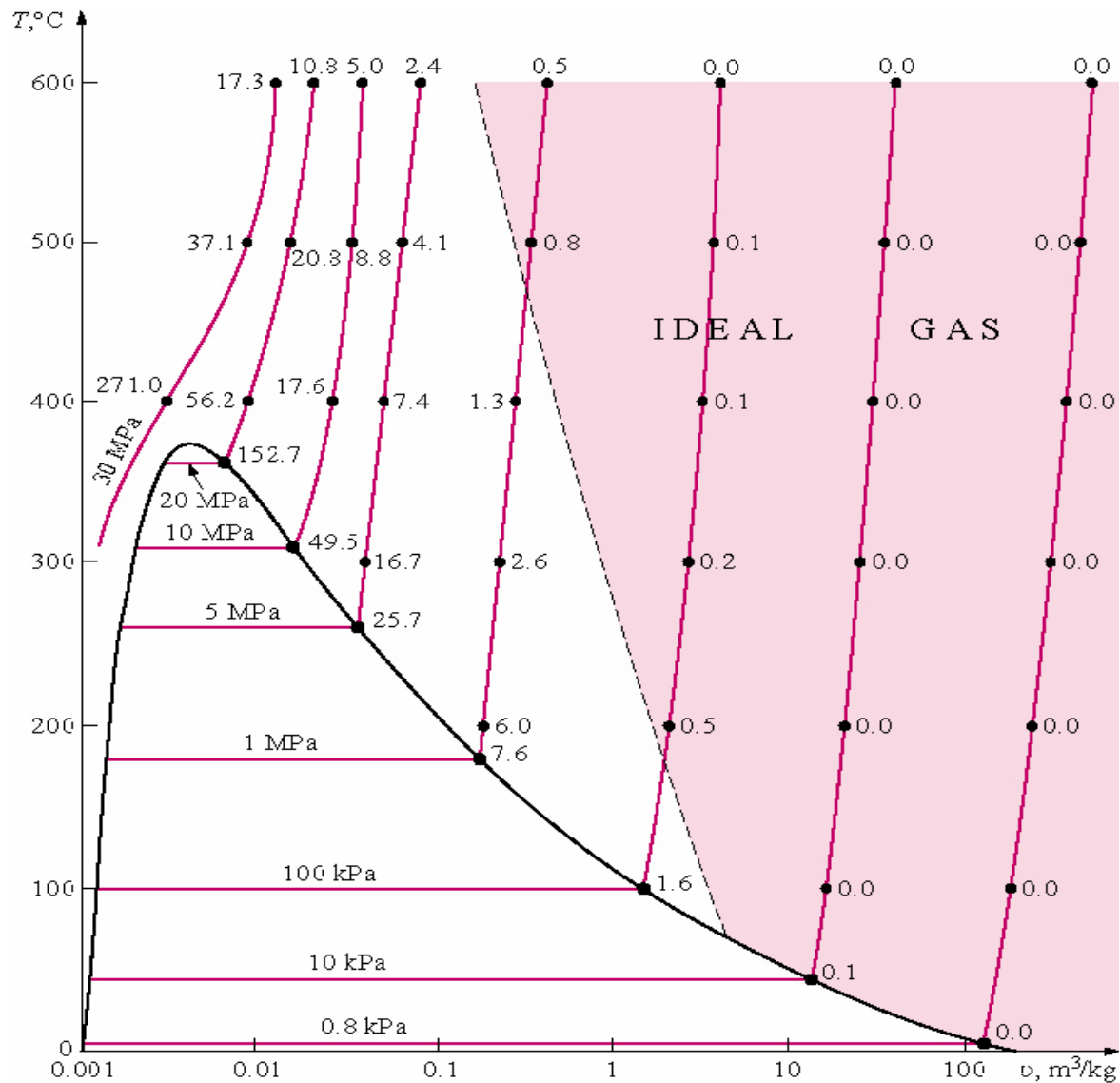


# *Question* .....

**Under what conditions is it appropriate to apply the ideal gas equation of state?**

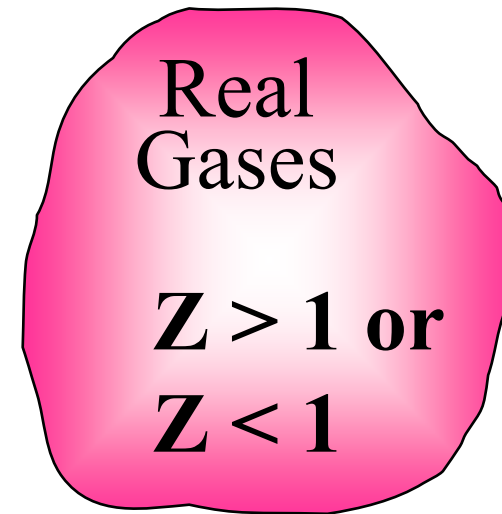
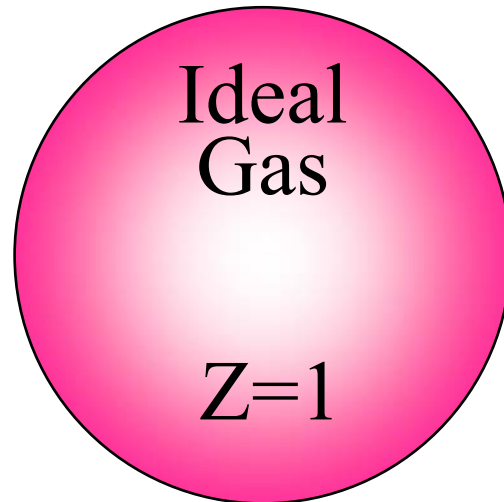
- ◆ **Good approximation for P-v-T behaviors of real gases at low densities (low pressure and high temperature).**
- ◆ **Air, nitrogen, oxygen, hydrogen, helium, argon, neon, carbon dioxide, .... (< 1% error).**

# Percent error for applying ideal gas equation of state to steam



# *Compressibility Factor*

- ◆ It accounts mainly for two things
  - Molecular structure
  - Intermolecular attractive forces



- ◆ What is it really doing?

# *Compressibility Factor*

- ♦ The deviation from ideal-gas behavior can be properly accounted for by using the *compressibility factor Z*, defined as

$$Z = \frac{P v}{RT} \quad \text{or} \quad Z = \frac{v_{actual}}{v_{ideal}}$$

- **Z represents the volume ratio or compressibility,**
- **Ideal Gas: Z =1;**
- **Real Gases: Z>1 or Z<1.**



# *Principle of corresponding states*

- ◆ The compressibility factor  $Z$  is approximately the same for all gases at the same *reduced temperature* and *reduced pressure*.

$$Z = Z(P_R, T_R) \text{ for all gases}$$



# *Reduced Pressure and Temperature*

$$P_R \equiv \frac{P}{P_{cr}} ; T_R \equiv \frac{T}{T_{cr}}$$

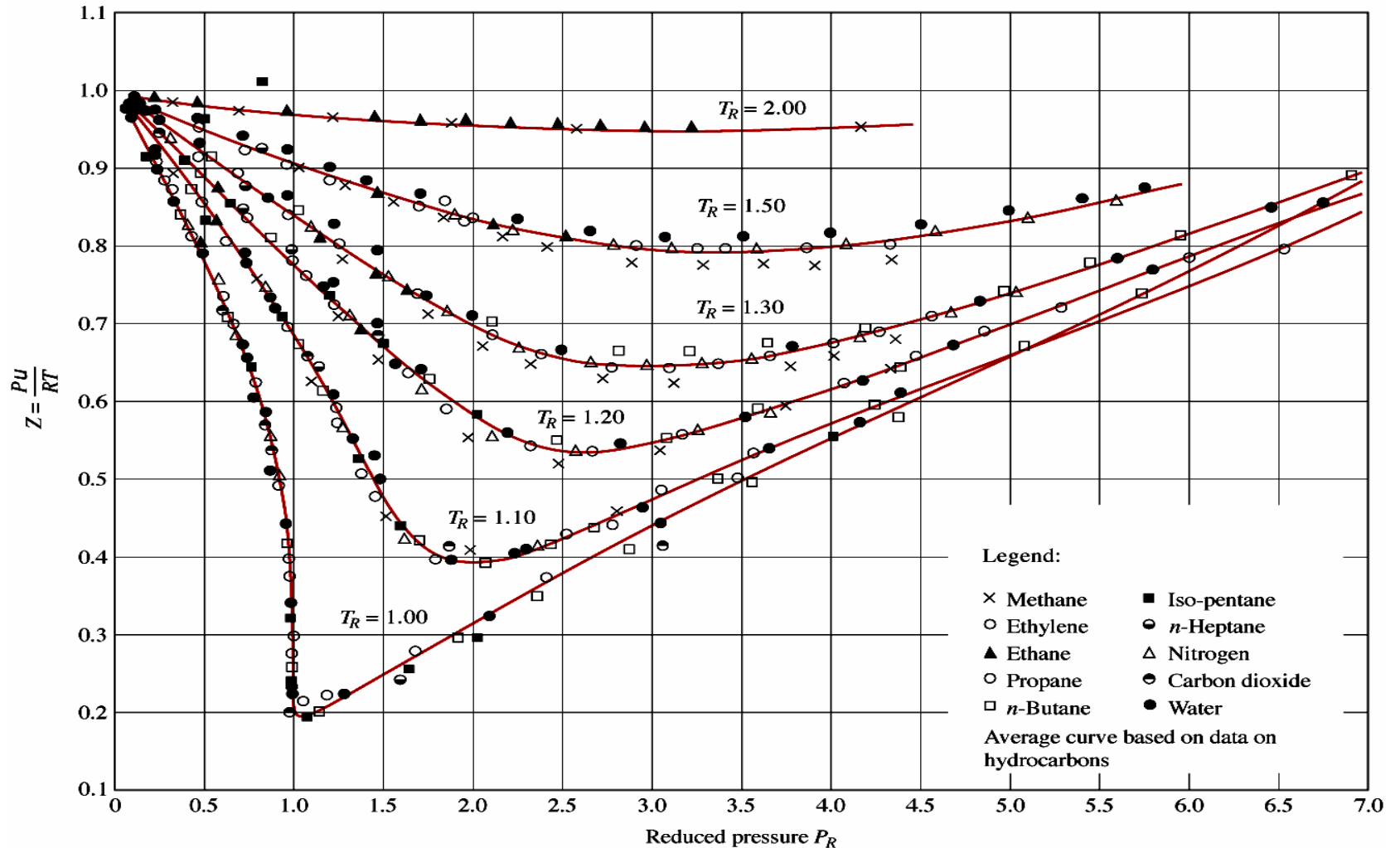
where:

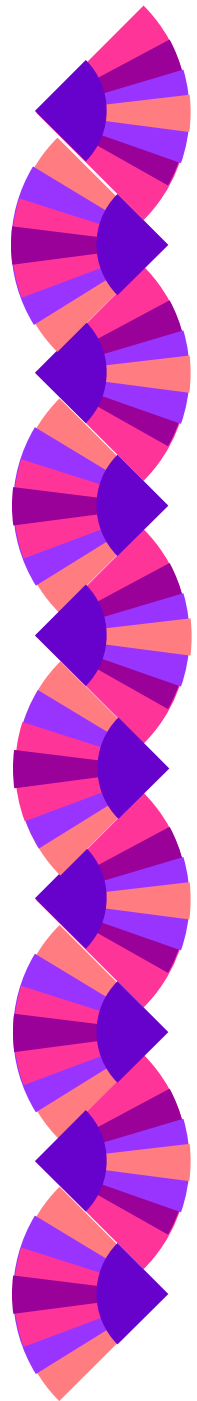
$P_R$  and  $T_R$  are reduced values.

$P_{cr}$  and  $T_{cr}$  are critical properties.

# Compressibility factor for ten substances

(applicable for all gases Table A-3)



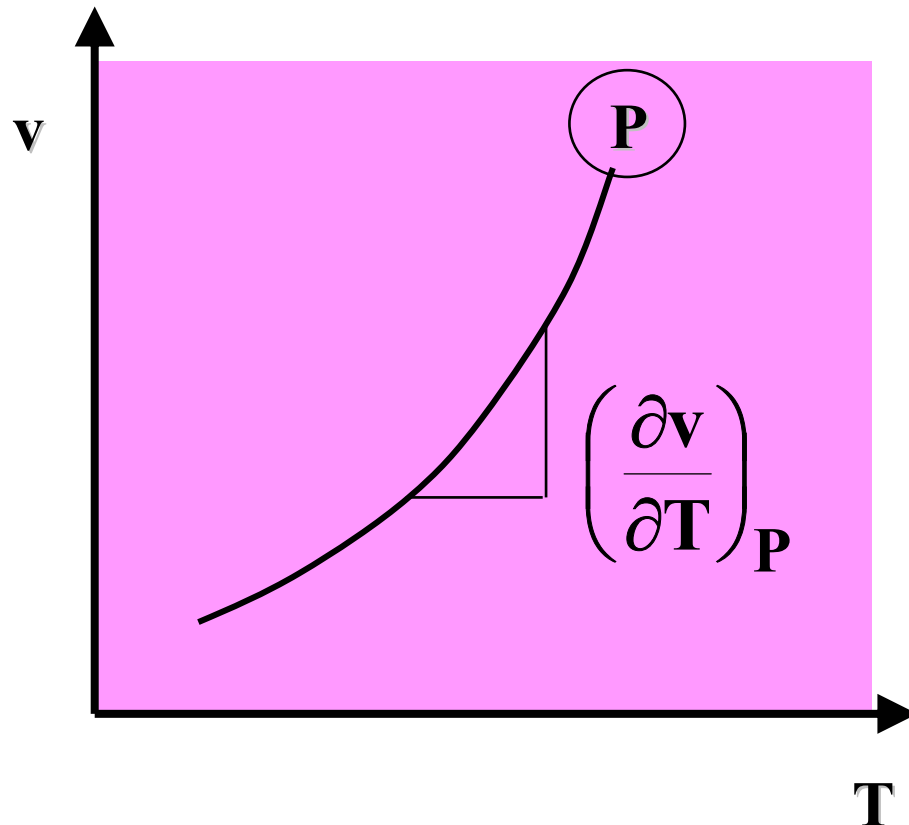


***OTHER  
THERMODYNAMIC  
PROPERTIES***



# *Other Thermodynamic Properties:*

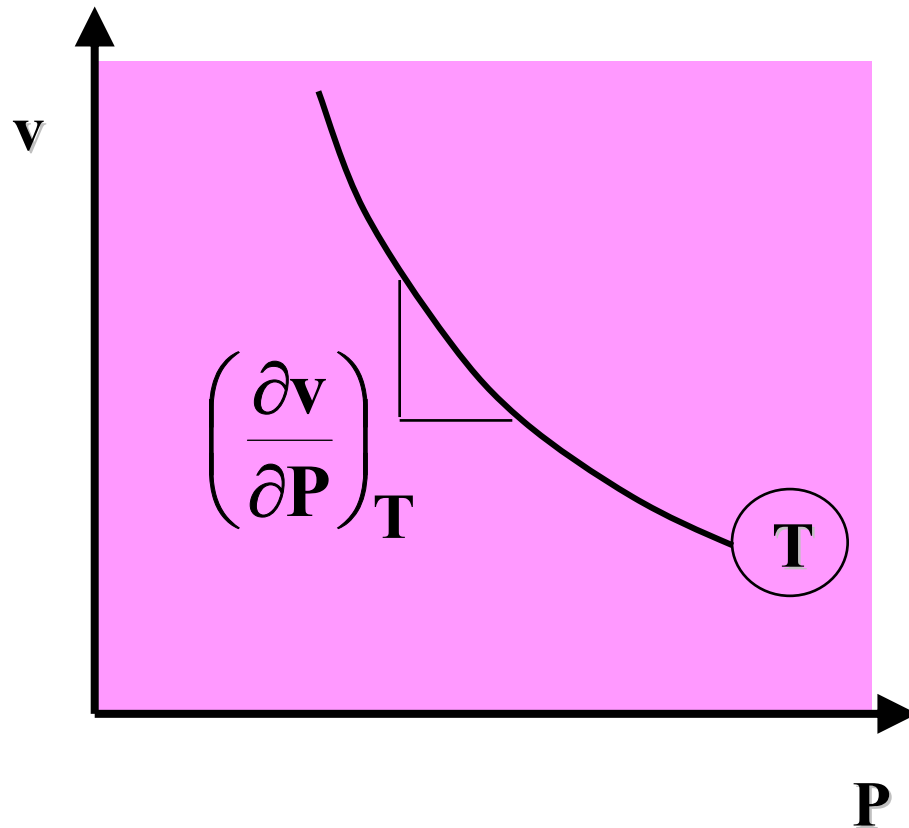
## *Isobaric (c. pressure) Coefficient*



$$\beta = \frac{1}{v} \left( \frac{\partial v}{\partial T} \right)_P > 0$$

For ideal gas,  
 $\beta = 1/T$

# *Other Thermodynamic Properties:* *Isothermal (c. temp) Coefficient*



$$\kappa = -\frac{1}{v} \left( \frac{\partial v}{\partial P} \right)_T > 0$$

For ideal gas,  
 $\kappa = 1/P$

# *Other Thermodynamic Properties:*

**We can think of the volume as being a function of pressure and temperature,  $v = v(P, T)$ . Hence infinitesimal differences in volume are expressed as infinitesimal differences in P and T, using  $\kappa$  and  $\beta$  coefficients**

$$dv = \left( \frac{\partial v}{\partial T} \right)_P dT + \left( \frac{\partial v}{\partial P} \right)_T dP \equiv \beta v dT - \kappa v dP$$

**For ideal gas:  $dv/v = dT/T - dP/P \rightarrow VP/T = \text{const!}$**

**For substances other than ideal gas, it can be approximated by:**

$$\text{Ln} \left( \frac{v}{v_0} \right) = \beta(T - T_0) - \kappa(P - P_0)$$

**If  $\kappa$  and  $\beta$  are nearly constant,**



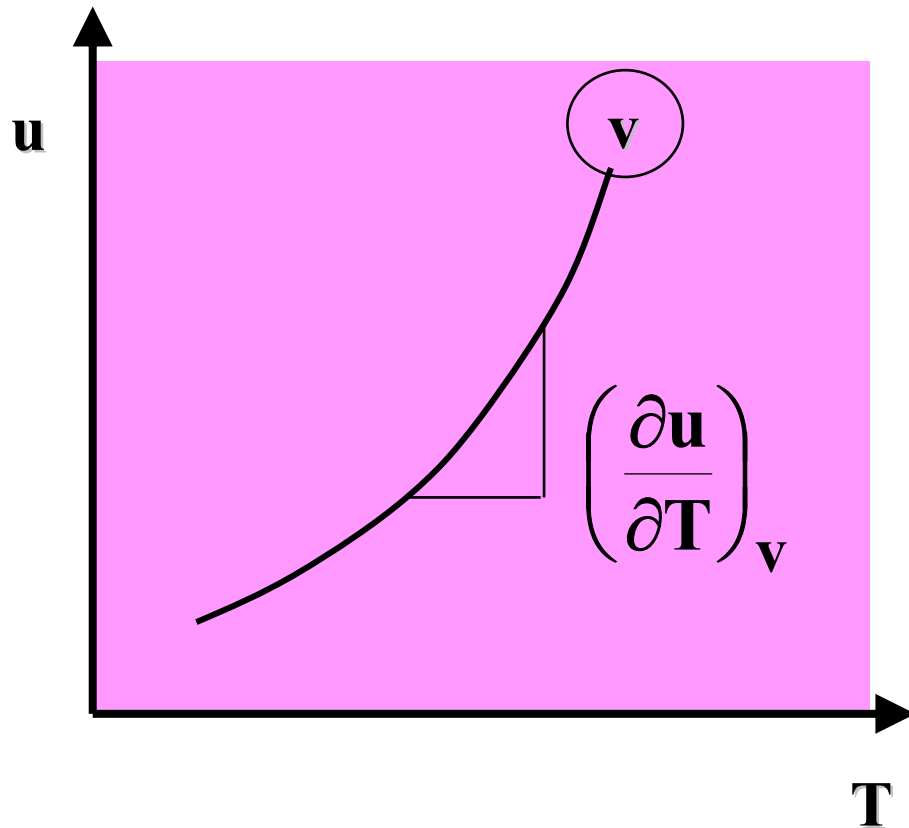
## *Other Thermodynamic Properties:*

$$\mathbf{u} = \mathbf{u}(T, \mathbf{v}) \quad - \mathbf{Internal\ Energy}$$

$$\mathbf{h} = \mathbf{h}(T, P) \equiv \mathbf{u} + P\mathbf{v} \quad - \mathbf{Enthalpy}$$

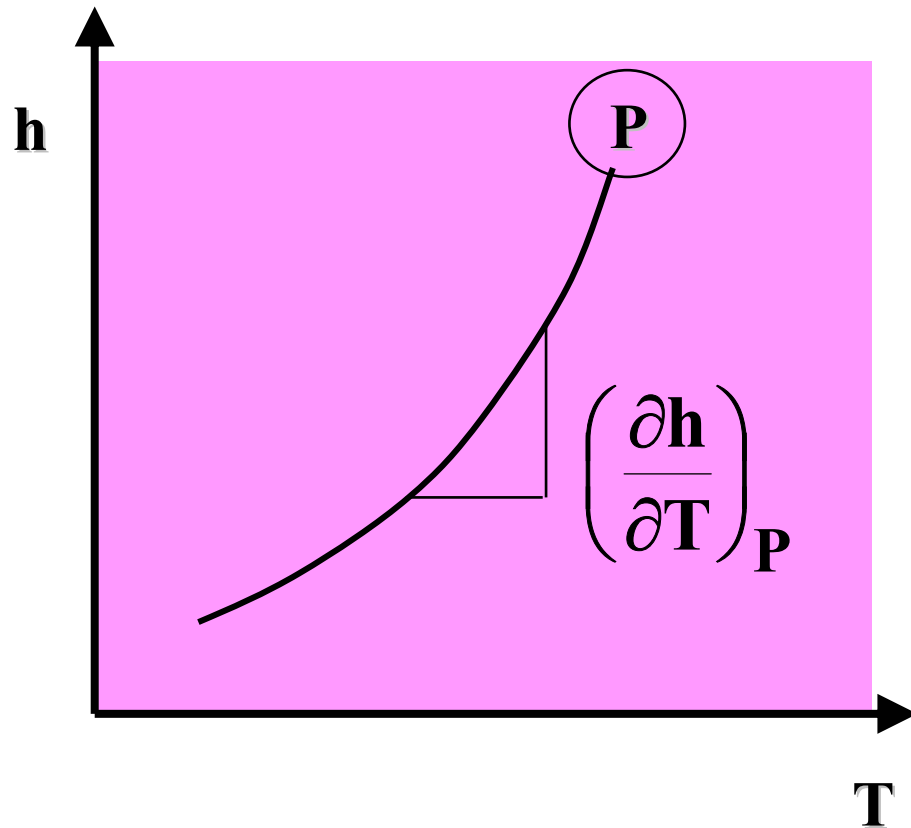
$$\mathbf{s} = \mathbf{s}(\mathbf{u}, \mathbf{v}) \quad - \mathbf{Entropy}$$

*Other Thermodynamic Properties:*  
*Specific Heat at Const. Volume*



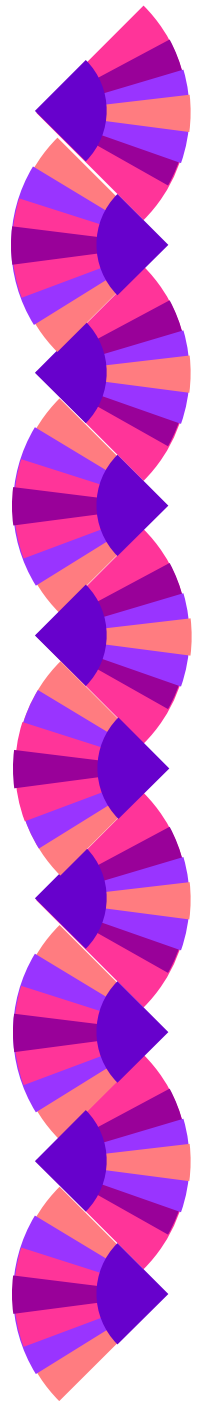
$$C_v = \left(\frac{\partial u}{\partial T}\right)_v > 0$$

*Other Thermodynamic Properties:*  
*Specific Heat at Const. Pressure*

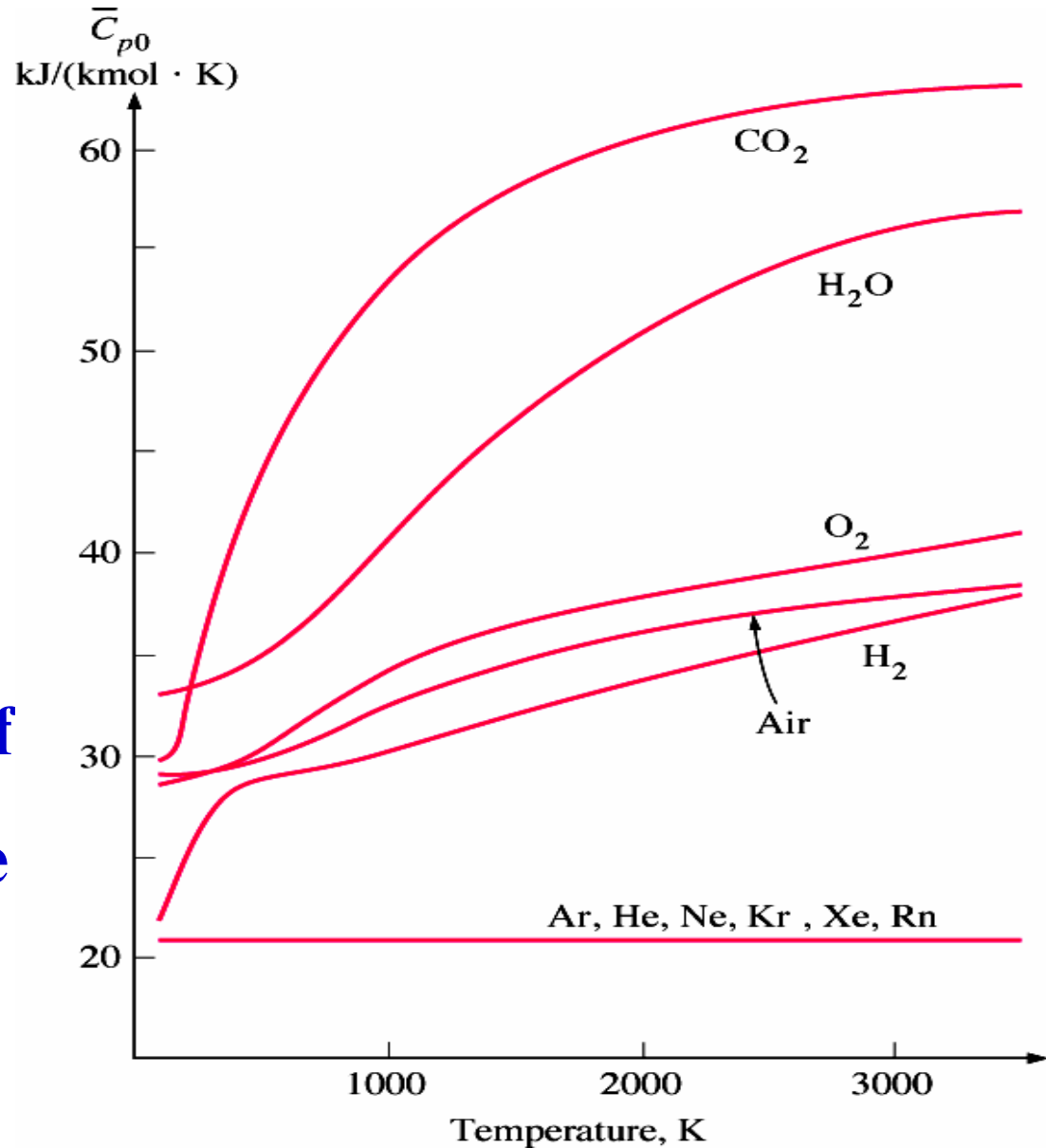


$$C_P = \left(\frac{\partial h}{\partial T}\right)_P > 0$$

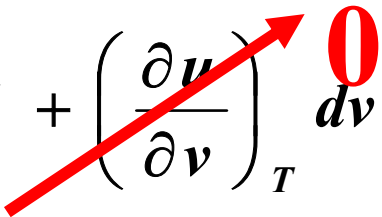
# Specific Heats for Some Gases



◆  $C_p = C_p(T)$   
a function of  
temperature



## *Ideal Gases: $u = u(T)$*

$$du = \left( \frac{\partial u}{\partial T} \right)_v dT + \left( \frac{\partial u}{\partial v} \right)_T dv$$


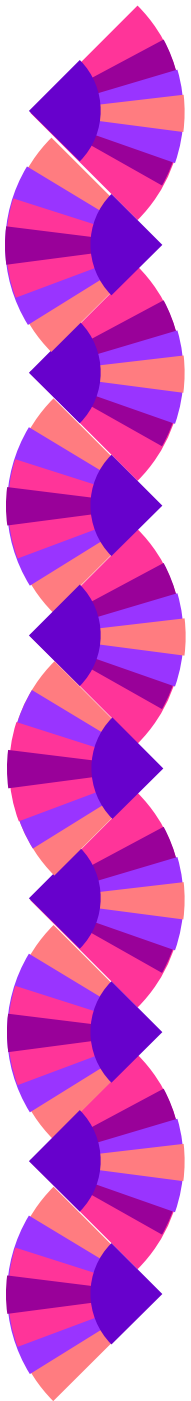
**Therefore,**

$$du = \left( \frac{\partial u}{\partial T} \right)_v dT = C_v(T) dT$$

**The internal energy change is:**

$$\Delta u = u_2 - u_1 = \int_{T_1}^{T_2} C_v(T) dT$$

As  $C_v$  changes with temperature it cannot be pulled out from the integral (*for general cases*).







## *Enthalpy for an Ideal Gas*

- ◆  **$h = u + Pv$**  where  **$Pv$**  can be replaced by  **$RT$**  because  **$Pv = RT$** .
- ◆ Therefore,  **$h = u + RT$**   $\Rightarrow$  since  **$u$**  is only a function of  **$T$** ,  **$R$**  is a constant, then  **$h$**  is also only a function of  **$T$** !
- ◆ so  **$h = h(T)$**



*Similarly, for a change in enthalpy for ideal gases:*

$$C_p = C_p(T) \quad \& \quad \left( \frac{\partial h}{\partial P} \right) \equiv 0$$

$$dh = C_p dT, \quad \text{and}$$

$$\Delta h = h_2 - h_1 = \int_{T_1}^{T_2} C_p(T) dT$$

# *Summary: Ideal Gases*

- ♦ For ideal gases  $u$ ,  $h$ ,  $C_v$ , and  $C_p$  are functions of **temperature** alone.
- ♦ For ideal gases,  $C_v$  and  $C_p$  are written in terms of ordinary differentials as

$$C_v = \left( \frac{du}{dT} \right)_{ideal\ gas} ; C_p = \left( \frac{dh}{dT} \right)_{ideal\ gas}$$



# *Three Ways to Calculate $\Delta u$ and $\Delta h$*

- ◆  $\Delta u = u_2 - u_1$  (table)

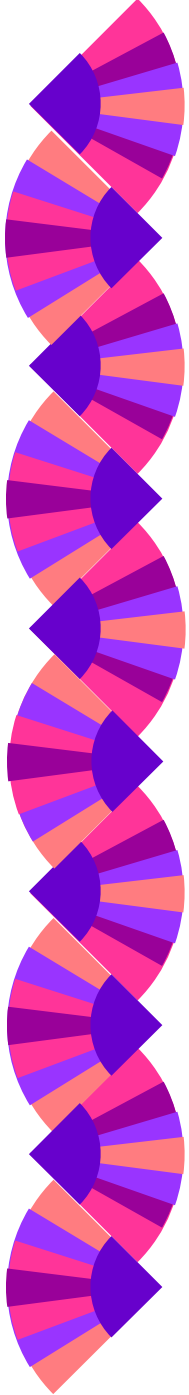
- ◆  $\Delta u = \int_1^2 C_v(T) dT$

- ◆  $\Delta u = C_{v,av} \Delta T$

- ◆  $\Delta h = h_2 - h_1$  (table)

- ◆  $\Delta h = \int_1^2 C_p(T) dT$

- ◆  $\Delta h = C_{p,av} \Delta T$

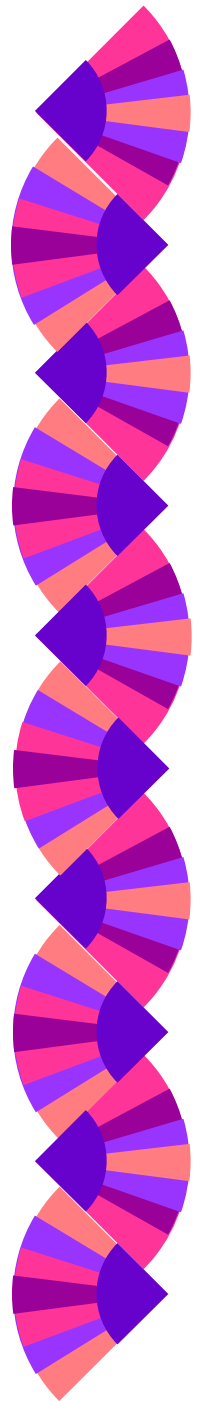


***Problem 3.31 – N<sub>2</sub> is heated from 373K to 1773K at constant pressure. Evaluate the specific enthalpy change (KJ per kg).***

♦  $\Delta h \approx C_{p,300k} \Delta T = \text{(tab. A-7)}$   
 $1,0416 \times (1773 - 373) = 1457 \text{ kJ/kg}$

♦  $\Delta h = \int_1^2 C_p(T) dT$  **tab A-5**,  $\theta = t(\text{K})/100$   
 $C_p = 39,060 - 512,79\theta^{-1.5} + 1072,7\theta^{-2} - 820,40\theta^{-3}$   
 $\Delta h = 1635.98 \text{ kJ/kg}$

***There is an error of 11% for using Cp constant when the temperature span is of 1400K!***



## ***$C_p \times C_v$ Relationship for an Ideal Gas,***

◆  $h = u + Pv = u + RT$

$$\frac{dh}{dT} = \frac{du}{dT} + R$$

$$C_p = C_v + R$$

$$\left[ \frac{\text{kJ}}{\text{kg} \cdot \text{K}} \right]$$



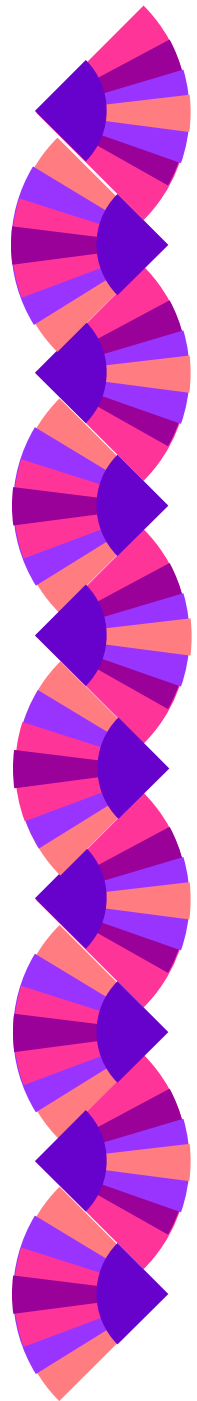
## Ratio of Specific Heats $\gamma$

$$\gamma \equiv \frac{C_p}{C_v} = \frac{C_p(T)}{C_v(T)} = \gamma(T)$$

For most gases  $\gamma$  is almost constant with temperature and equals to 1.4

$$\frac{C_p}{C_v} = 1 + \frac{R}{C_v} = \gamma > 1$$

$$C_v = \frac{R}{\gamma-1} \quad \text{and} \quad C_p = \frac{\gamma R}{\gamma-1}$$

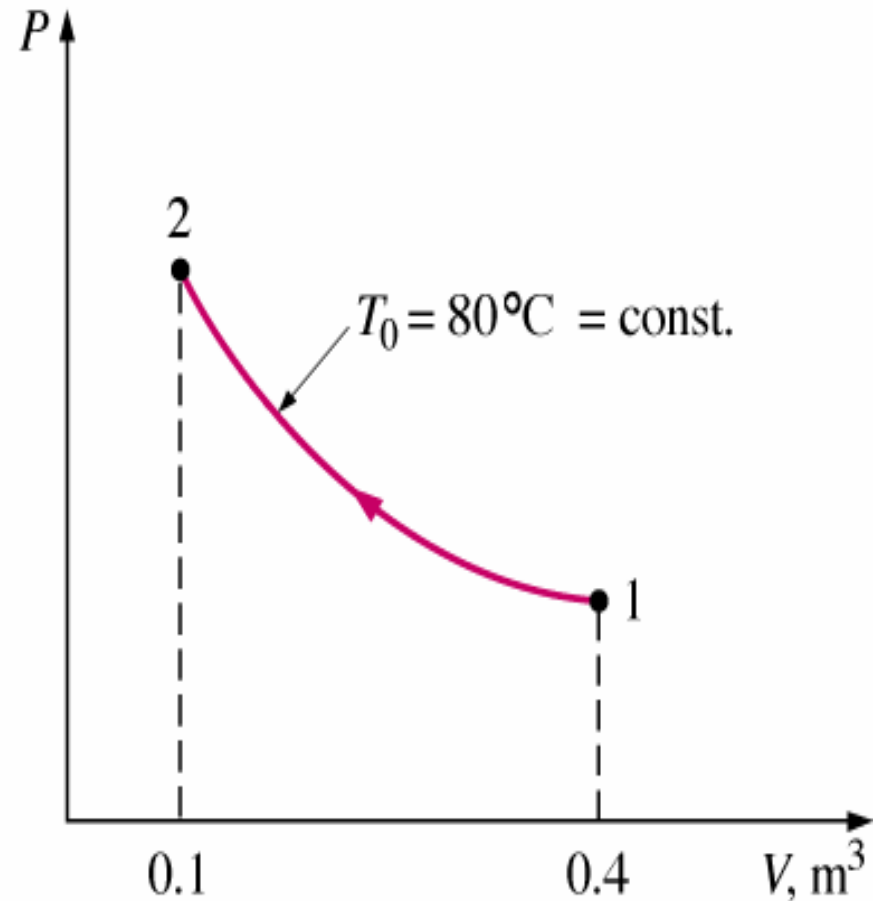
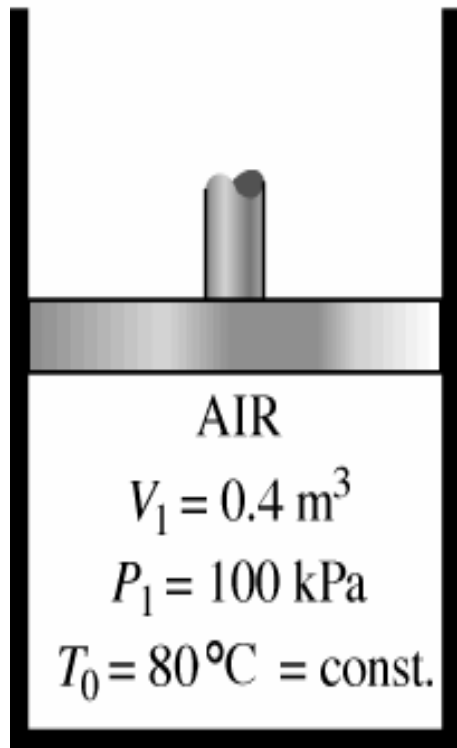


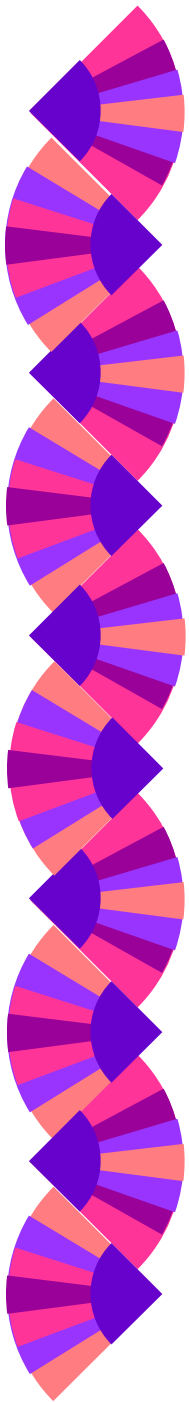
***ISOTHERMAL,  
POLITROPIC AND  
ADIABATIC PROCESSES  
FOR AN IDEAL GAS***



# Isothermal Process

- ◆ Ideal gas:  $PV = mRT = \text{constant}$





**For ideal gas,  $PV = mRT$**   
**We substitute into the**  
**integral**

$$W_b = \int_1^2 PdV = \int_1^2 \frac{mRT}{V} dV$$

**Collecting terms and integrating yields:**

$$W_b = mRT \int_1^2 \frac{dV}{V} = mRT \ln \left( \frac{V_2}{V_1} \right)$$

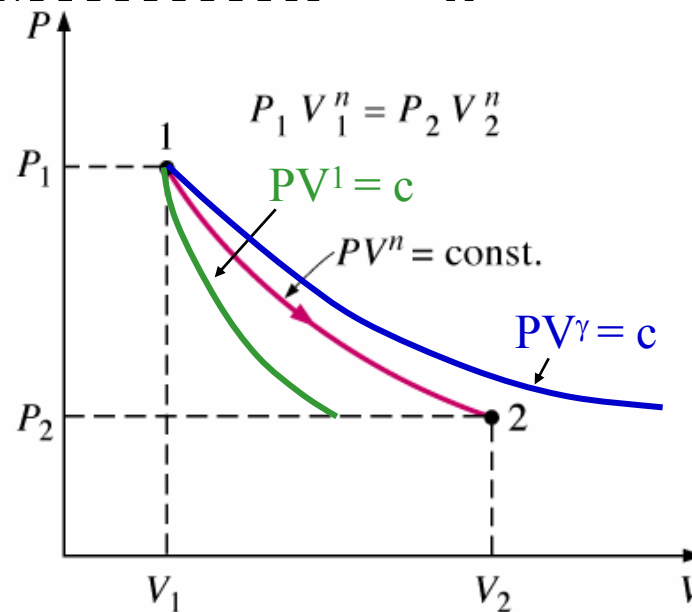
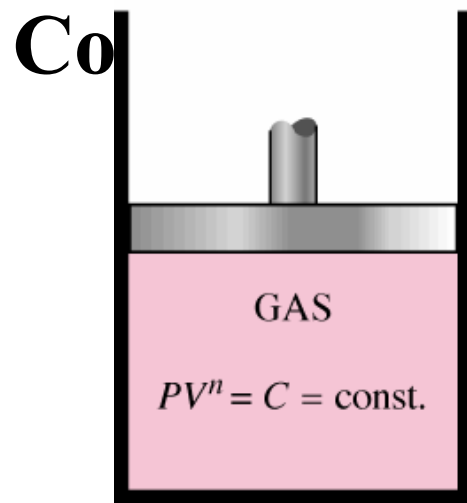
# Polytropic Process: $PV^n = C$ ; $1 \leq n \leq \gamma$

Isothermal .....  $n = 1$

Adiabatic ( $Q=0$ ).....  $n = \gamma = C_p/C_v$

*Others:*

Constant pressure.....  $n = 0$



The politropic lines are always to the right of the isothermal lines.

# P, v and T Relationship in a Polytropic Process, $1 \leq n \leq \infty$

- ◆ The path is described by:  $Pv^n = C$
- ◆ The ideal gas state equation is:  $Pv = RT$
- ◆ Combining these two expressions is possible to relate the initial to the final states

$$\left(\frac{P_2}{P_1}\right) = \left(\frac{V_1}{V_2}\right)^n \quad \left(\frac{T_2}{T_1}\right) = \left(\frac{V_1}{V_2}\right)^{n-1}$$

$$\left(\frac{T_2}{T_1}\right) = \left(\frac{P_2}{P_1}\right)^{\frac{n-1}{n}}$$



# Boundary work for a gas which obeys the polytropic equation

$$\begin{aligned} W_b &= \int_1^2 P dV = c \int_1^2 \frac{dV}{V^n} \\ &= c \left[ \frac{V^{1-n}}{1-n} \right]_{V_1}^{V_2} = c \left[ \frac{V_2^{1-n} - V_1^{1-n}}{1-n} \right] \quad \text{para } (n \neq 1) \end{aligned}$$

Where **C** stands for the constant  $Pv^n = C$



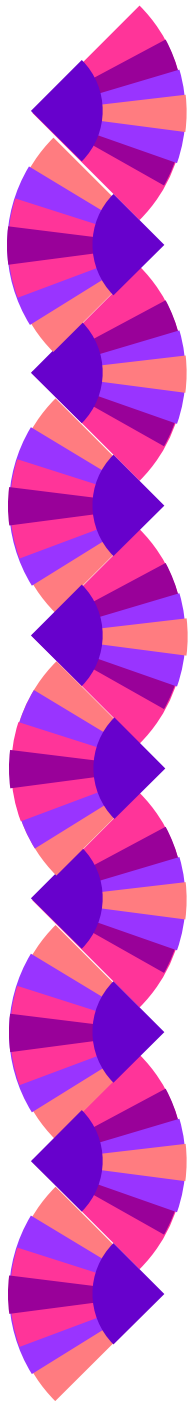
**We can further simplify**

**The constant  $c = P_1 V_1^n = P_2 V_2^n$**

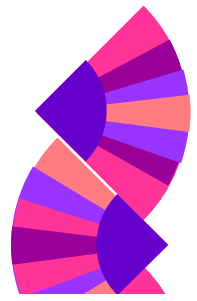
$$W_b = \frac{P_2 V_2^n (V_2^{1-n}) - P_1 V_1^n (V_1^{1-n})}{1-n}$$
$$= \frac{P_2 V_2 - P_1 V_1}{1-n}, \quad n \neq 1$$

# Summary for Polytropic Process

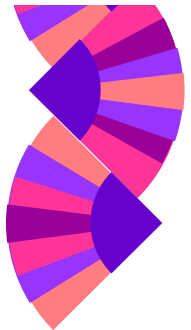
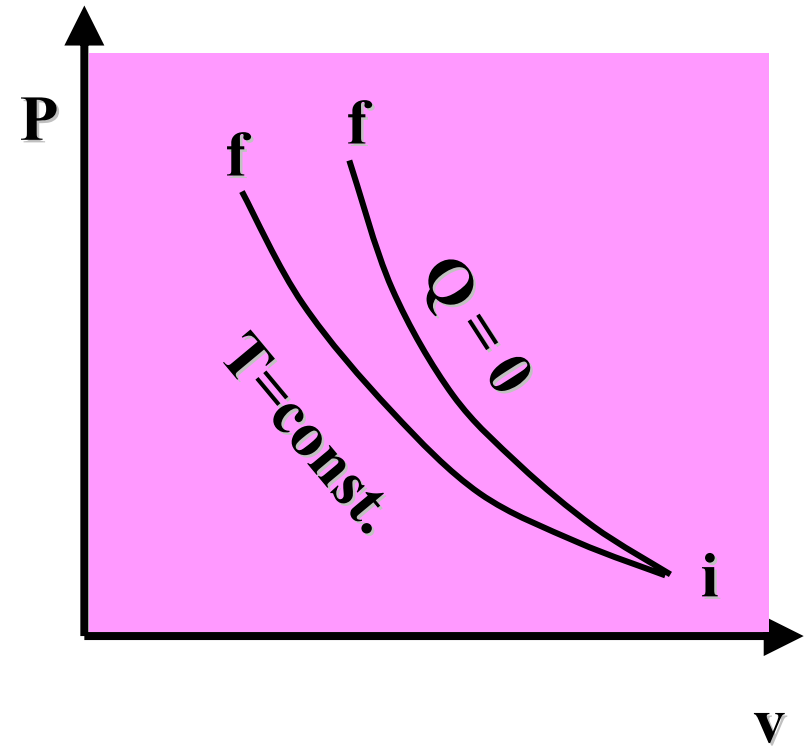
$$\begin{aligned}W_b &= \int_1^2 P dV = \int_1^2 \frac{c}{V^n} dV \\&= \frac{P_2 V_2 - P_1 V_1}{1 - n}, \quad n \neq 1 \\&= PV \ln \left( \frac{V_2}{V_1} \right), \quad n = 1\end{aligned}$$



# *Ideal Gas Adiabatic Process and Reversible Work*



- ◆ Why  $Pv^\gamma = C$  represents a process where the volume is expanding or contracting without heat flux,  $Q = 0$ ?





# *Ideal Gas Adiabatic Process and Reversible Work (cont)*

**First Law:**

$$\underbrace{dQ}_{=0} - \underbrace{dW}_{PdV} = \underbrace{dU}_{MC_V dT}$$

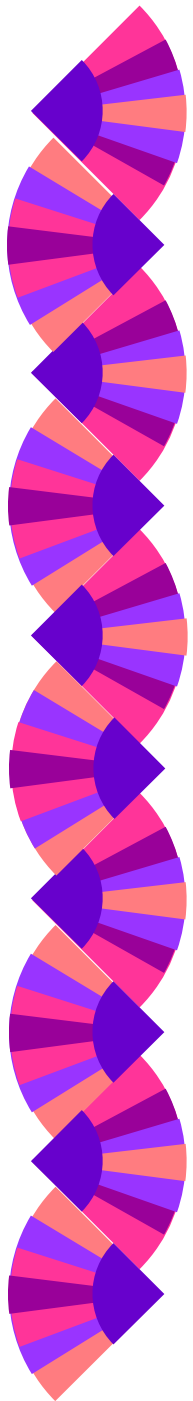
**Substituting**  
**P = MRT/V**

$$\left(\frac{dV}{V}\right) = - \frac{C_V}{\underbrace{R}_{(1-\gamma)^{-1}}} \left(\frac{dT}{T}\right)$$

**Integrating from**  
**(1) to (2)**

$$\left(\frac{T_2}{T_1}\right) = \left(\frac{V_2}{V_1}\right)^{(1-\gamma)} \Rightarrow \left(\frac{P_2}{P_1}\right) = \left(\frac{V_1}{V_2}\right)^\gamma$$

Which are the polytropic relations seen before!

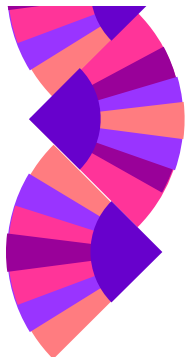




# Exercise 3-30

**Air is compressed reversibly and adiabatically from a pressure of 0.1 MPa and a temperature of 20°C to a pressure of 1.0 MPa.**

- a) Find the air temperature after the compression**
- b) What is the density ratio (after to before compression)**
- c) How much work is done in compressing 2 kg of air?**
- d) How much power is required to compress 2kg per second of air?**





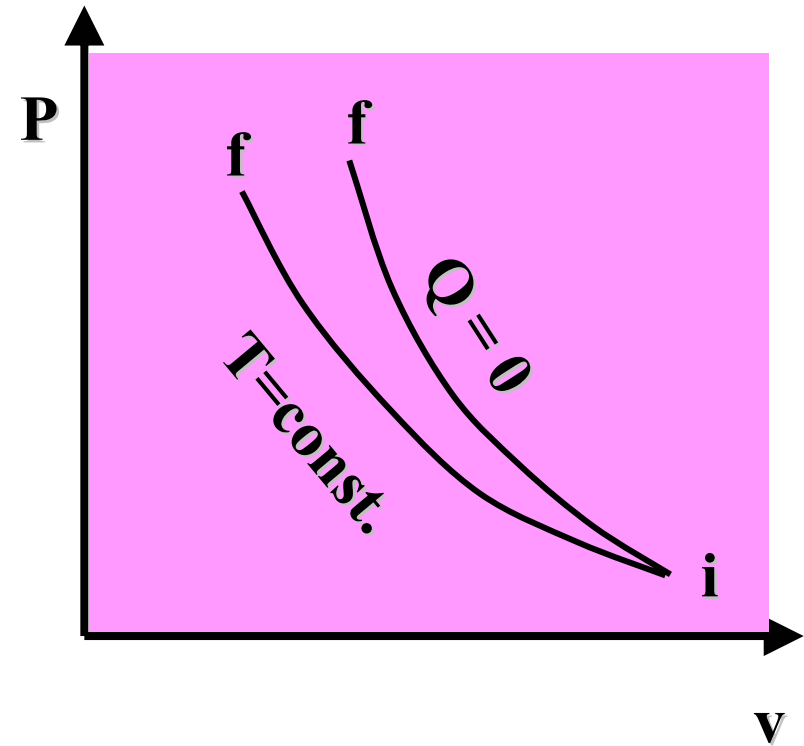
# Solution - page 1

- ◆ In a reversible and adiabatic process  $P$ ,  $T$  and  $v$  follows:

$$\left(\frac{T_2}{T_1}\right) = \left(\frac{V_1}{V_2}\right)^{(\gamma-1)}$$

$$\left(\frac{P_2}{P_1}\right) = \left(\frac{V_1}{V_2}\right)^\gamma$$

$$\left(\frac{T_2}{T_1}\right) = \left(\frac{P_2}{P_1}\right)^{(\gamma-1)/\gamma}$$



# Solution - page 2

## Part a)

The temperature after compression is

$$T_2 = T_1 \cdot \left( \frac{P_2}{P_1} \right)^{(\gamma-1)/\gamma} \rightarrow T_2 = 293 \cdot \left( \frac{1}{0.1} \right)^{0.4/1.4} = 566\text{K} \quad (293\text{oC})$$

## Part b)

The density ratio is

$$\left( \frac{V_2}{V_1} \right) \equiv \left( \frac{\rho_1}{\rho_2} \right) = \left( \frac{P_2}{P_1} \right)^{-1/\gamma} \rightarrow \left( \frac{\rho_2}{\rho_1} \right) = \left( \frac{1}{0.1} \right)^{1/1.4} = 5.179$$

# Solution - page 3

**Part c)**

**The reversible work:**

$$\begin{aligned} W_{\text{REV}} &= -\frac{[(PV)_2 - (PV)_1]}{(\gamma - 1)} \equiv \frac{M \cdot R(T_2 - T_1)}{(\gamma - 1)} = \\ &= \frac{2 \cdot 287 \cdot (566 - 293)}{0.4} = 391 \text{KJ} \end{aligned}$$

**Part d)**

**The power is:**

$$P = \frac{dW}{dt} = -\frac{\dot{M}R(T_2 - T_1)}{(\gamma - 1)} = 391 \text{KW}$$



*Exercícios – Capítulo 3*  
*Propriedades das Substâncias*  
*Puras*

**Exercícios Propostos: 3.6 / 3.9 / 3.12 /  
3.16 / 3.21 / 3.22 / 3.26 / 3.30 / 3.32 / 3.34**

**Team Play: 3.1 / 3.2 / 3.4**