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Particle image velocimetry in a centrifugal pump: Influence of walls on the flow at different axial positions

For almost a century, humans have relied on centrifugal pumps for the transport of low-viscous fluids in commercial, agricultural, and industrial activities. Details of the fluid flow in impellers often influence the overall performance of the centrifugal pump, and may explain unstable and inefficient operations taking place sometimes. However, most studies in the literature were devoted to understanding the flow in the mid-axial position of the impeller, only a few focusing their analysis on regions closer to solid walls. This paper aims at studying the water flow on the vicinity of the front and rear covers (shroud and hub) of a radial impeller to address the influence of these walls on the fluid dynamics. For that, experiments using particle image velocimetry (PIV) were conducted in a transparent pump at three different axial planes, and the PIV images were processed for obtaining the average velocity fields and profiles, as well as turbulence levels. Our results suggest that: (i) significant angular deviations are observed when the velocity vectors on the peripheral planes are compared with those on the central plane; (ii) the velocity profiles close to the border are similar to those in the middle, but the magnitudes are lower close to the hub than to the shroud; (iii) the turbulent kinetic energy on the periphery is up to eight times greater than that measured at the center. Our results bring new insights that can help proposing mathematical models and improving the design of new impellers. A database and technical drawings of the centrifugal pump are also available in this paper, so that other researchers can perform numerical simulations and validate them against experimental data.

Keywords: Particle Image Velocimetry, Centrifugal Pump, Velocity Field

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1 INTRODUCTION

2 Many human activities depend on centrifugal pumps, which are frequently used for the horizontal and vertical transport
3 of liquids in single and multiphase flows. Basically, these devices consist of a rotating component (impeller) that supplies
4 energy to the fluid, and a stationary part (diffuser) that converts part of the kinetic energy into pressure energy [1, 2]. In
5 the particular case of oil production, an electrical submersible pump (ESP) is normally installed inside the oil well to act
6 as an artificial lift method [3, 4].

7 From the increasing demands of energy saving and long-term operation, the flow structures formed inside impellers and
8 their influence on the pump efficiency have become constant topics of research in the field of fluid mechanics. At off-design
9 conditions, for instance, the flow inside a pump is usually complex due to the presence of unstable structures, engendering
10 high temporal and spatial velocity gradients, regions of intense vorticity and turbulence levels, as well as zones with flow
11 separation, re-circulation and reverse flow. These characteristics directly affect the pump performance and may be a cause
12 of undesirable energy losses [5, 6].

13 The particle image velocimetry (PIV) technique is non-intrusive, i.e., it does not disturb or modify the flow being
14 investigated [7], therefore the method has proved to be advantageous for carrying out measurements on rotating devices
15 such as stirred tanks, turbines, and pumps. In this context, many researchers worldwide have been using the PIV as a
16 measuring approach to investigate the internal flow in pumps. Paone et al. [8] and Dong et al. [9, 10] were the first to
17 study single-phase flows in a centrifugal pump using PIV. They obtained velocity fields that were analyzed with focus on
18 vorticity and turbulence for describing internal flow patterns and interactions between the volute tongue and the impeller.

19 As technology applied to lasers, cameras, and computers improved, digital PIV methods became accessible to a larger
20 number of researchers. For instance, Sinha and Katz [11] and Sinha et al. [12] used PIV to investigate the wakes generated
21 by impeller blades, diffuser vanes, and unsteady separation phenomena in centrifugal pumps. Pedersen et al. [13] measured
22 the instantaneous and average velocity fields in a pump working at the design point and low flow rates, and compared
23 the results with data obtained with laser Doppler velocimetry (LDV). Recently, Keller et al. [14] used a PIV system to
24 analyze the unsteady flow field near the volute tongue of a centrifugal pump. For different percentages of the flow rate
25 corresponding to the best efficiency point (BEP), they found that vorticity sheets appear on the surfaces of the impeller
26 blades as a consequence of velocity gradients. Besides, regions with a high turbulent kinetic energy production level were
27 detected in the trailing edge of the blades. These are examples showing that significant attention was devoted to investigate
28 the internal flow in pumps, over the last decades [15–18].

29 Presently, the studies that rely on PIV techniques to investigate centrifugal pumps are mainly focused on the influence
30 of the flow patterns on hydraulic performance [19–21], presence of vortices or rotating stalls [22–25], and characterization
31 of turbulence levels through the identification of coherent flow structures [26]. These studies, however, cannot evaluate the
32 influence of secondary flows on the general characteristics of the pump's internal flow. Although there are a few studies
33 [14, 27–29] that use the stereoscopic version of the PIV method (2D3C-PIV) to assess the entire volume of fluid in the
34 impeller, most publications available in literature actually rely on the standard two-dimensional PIV (2D2C-PIV) with thin
35 laser sheets that illuminate areas - instead of volumes - of the pump stage. The common configuration of 2D2C-PIV

enables the measurements on a plane, so that the velocity is computed in the x and y directions, but not in the z direction. This axial position, z , is usually set up in the middle of the impeller height, far from the solid walls.

This approach was adopted in our previous paper, Perissinotto et al. [5], which described a methodology for applying the 2D2C-PIV in a pump impeller, providing results for the midplane only. Guided by the opportunity to expand our study, we propose the present manuscript, which aims to investigate the flow in different axial positions of the same impeller. This strategy of performing measurements at different axial planes of radial impellers has been recently adopted by Zhang et al. [30] and Ofuchi et al. [31]. Nevertheless, assessing the hub-to-shroud flow behavior is actually a difficulty task, and there is still room to expand this type of analysis in literature. In this sense, we carried out PIV measurements at three axial positions to assess the influence of solid walls on the flow characteristics. In the tests, the laser plane was positioned successively at the impeller center (equidistant from shroud and hub), next to the bottom cover (hub), and next to the top cover (shroud) of the impeller, and single-phase water flows were imposed for different pump rotations.

By analyzing these 3 sets of results (upper, central, lower planes) and comparing them with our previous results (extracted from [5]), we can state that: (i) significant angular deviations are observed when the velocity vectors on the peripheral planes are compared with those on the central plane; (ii) the velocity profiles close to the border are similar to those in the middle, but the magnitudes are lower close to the hub than to the shroud; and (iii) the turbulent kinetic energy on the periphery is up to eight times greater than that measured at the center. These observations help improve the understanding of fluid dynamics within centrifugal pumps, as well as the relationship between the flow behavior and pump performance.

In the following, Section 2 presents the experimental setup, Section 3 discusses the results, and Section 4 finally summarizes the main conclusions.

2 EXPERIMENTAL SETUP

This section describes the experimental setup of the current study, including the test facility (subsection 2.1), PIV system (subsection 2.2), data processing and computations (subsection 2.3).

2.1 Test bench with transparent pump. The test bench used in this work is the same described in Perissinotto et al. [5]. The layout of the experimental facility is depicted in Fig. 1. As can be seen, the setup is essentially composed of a water flow line with a tank, a re-circulation pump, a booster pump, a transparent pump, and a PIV system. Instruments with relative uncertainties lower than 0.5% measure the water flow rate (Q_w) and temperature (T), pressure in the suction point of the transparent pump (P), pressure increment generated by this same pump (ΔP), and rotational speed of its impeller (N). The analog output signals from these instruments are acquired and the measured data monitored and stored by a supervisory control program. During the experiments, Q_w and P are controlled by setting the rotational speed of the booster pump and adjusting the opening or closing level of the valve assembled at the discharge point of the transparent pump. Furthermore, T is kept constant at approximately 25°C by a heat exchanger and thermochiller system.

The transparent centrifugal pump installed in the test bench was also used in Perissinotto et al. [5] and consists of an impeller (closed radial type), a vaneless diffuser (volute type), and other parts such as body, shaft, intake and discharge ports. Details of the design and manufacture are available in Perissinotto et al. [32]. To enable optical measurements

70 using the PIV method, the shroud and blades of the impeller are made of acrylic (or plexiglass), while the hub is made
71 of matte black aluminum. The transparent parts enable the laser sheets to enter and exit the pump while the dark surfaces
72 help reduce undesirable reflections and enhance contrast in the images.

73 The impeller has a radial geometry based on a real electrical submersible pump (ESP), P23 model, *Centrilift 538* series,
74 manufactured by *Baker Hughes*. It has seven channels of a constant height $h_{imp} = 6$ mm, inner diameter $d_1 = 44$ mm, and
75 outer diameter $d_2 = 110$ mm, so that the aspect ratio is low, $h_{imp}/d_2 \approx 5\%$. The volute spiral has a radius r_{vol} that varies
76 as a function of the angle, so that $58.5 < r_{vol} < 92.0$ mm. This component has a rectangular cross section with a constant
77 height $h_{vol} = 11$ mm in the z direction [32]. A photograph of the visualization stage of the transparent pump is displayed
78 in Fig. 2, which highlights the closed radial impeller.

79 The liquid enters the pump through four intake ports located at the pump body. Then, the fluid reaches the suction eye in
80 the impeller inlet, traverses the channels, and leaves the impeller to be collected by the volute. The liquid is finally directed
81 to the discharge port, placed at the end of the spiral [5]. Subsection 3.4 contains technical drawings of the transparent
82 pump.

83 **2.2 Test matrix and PIV setup.** The test matrix for the present study contains six operational conditions, as listed in
84 Tab. 1: two rotational speeds (N) and three flow rates (Q_w) referenced as percentages of the flow rate corresponding to
85 the BEP (Q_{BEP}). This experimental campaign repeats the conditions analyzed by Perissinotto et al. [5] and then allows
86 a more adequate comparison between results. Flow visualization was carried out using a two-dimensional PIV technique,
87 with a system already described in [5]. This system is a time-resolved equipment assembled by *Dantec Dynamics*® that
88 provides an energy of 30 mJ per pulse when working at 1 kHz. In this paper, the laser was set to work at 200 Hz.

89 As a complement to the information available in Tab. 1, the head curves of the transparent pump operating at both
90 rotational speeds are shown in Fig. 3. The pump head is defined as $H = \Delta P/\rho g$ [33], where ΔP is the pressure generated
91 by the pump, ρ is the water density, g is the modulus of the gravitational acceleration \vec{g} . The flow rates investigated in this
92 study are highlighted in these curves, i.e., conditions close to shut-off, BEP, and open-flow. Curves of dimensionless head
93 are available in [5] together with torque measurements and efficiency calculations. However, for the reader's convenience,
94 efficiency curves are also plotted in Fig. 3 to supplement the pump performance information. As it is possible to observe,
95 the efficiency is lower than 10%, as a consequence of the energy dissipation occurring in mechanical elements assembled
96 on the pump shaft, such as bearings and seals, which compose the transmission system of the motor-pump set [5].

97 During the PIV tests, microspheres of acrylic doped with rhodamine-B dye, with diameters from 20 to 50 μm , are added
98 to water to serve as fluorescent tracers in the scope of laser-induced fluorescence (LIF). However, for low flow rates, these
99 particles tend to deposit at the bottom of the pump stage, over the course of the test. Thus, $Q_w = 0.15$ m^3/h is the lowest
100 possible flow rate for acquiring adequate images, being considered here as the shut-off point. In addition, the time between
101 two consecutive pulses of the laser generator must be set within the range from 300 to 800 μs to limit the maximum
102 displacement of the particles to 10 pixels in the images [5].

103 Each experimental condition demands the acquisition of 500 pairs of images in the PIV measurements. A high-speed
104 camera with a spatial resolution of 2560 px \times 1600 px is used for capturing these images, having a 0.1 mm/px relation for

105 the the fields of view used in this work. The acquisitions are performed only when the impeller reaches a predetermined
106 angular position, in which the tip of the volute tongue becomes vertically aligned with the tip of a predefined impeller
107 blade. For that, a triggering system is used, consisting of a synchronizer and a rotary encoder assembled on the pump shaft
108 [5] (see Fig. 1).

109 The illumination planes are positioned perpendicularly to the camera lens, in a configuration that illuminates the entire
110 pump stage (x and y direction) at either of three locations along the impeller height (z direction): (i) on the central plane
111 (equidistant from solid walls); (ii) next to the internal surface of the impeller hub (displaced -3 mm from the center); and
112 (iii) next to the internal surface of the impeller shroud (displaced $+3$ mm from the center), as observed in Fig. 4. The
113 use of two planes in opposite directions, entering from both sides of the pump, is intended to make the lighting more
114 effective. Eventual shadows caused by the shaft and distortions caused by the blades are mitigated through this double-plane
115 configuration.

116 Before starting each experiment, the thickness of the laser sheet is set to approximately 1 mm, and the axial position
117 adjusted and confirmed with a dial gauge indicator. The variations in the axial positions of the planes, in the configurations
118 (ii) and (iii), are the most significant difference between the experiments carried out in the present manuscript and those
119 performed before, in Perissinotto et al. [5].

120 **2.3 Data processing and calculations.** When the experiment is completed, a pre-processing code is applied to the 500
121 raw images in order to mask out solid walls, including the blades, where the fluid is not present. Then, a numerical code
122 based on Liu et al. [34] is applied to remove the angular displacement of the impeller between consecutive images. This
123 procedure ensures that the location of the blades is exactly the same in both frames of each pair of images. The images
124 are then sent to a cross-correlation routine carried out on the *DynamicStudio 7.4* software. An adaptive PIV method, as
125 reported by Scarano and Riethmuller [35], with initial and final interrogation regions of 64 px and 32 px and 25% overlap,
126 is responsible for calculating the velocity vectors.

127 The result is a set of 500 instantaneous velocity fields, and by using an in-house software, these fields are finally converted
128 into a single field of phase-ensemble averaged velocities, as minutely described in Perissinotto et al. [5]. (This amount of
129 500 images is sufficient to provide fields with satisfactory quality. From preliminary tests, we verified that convergence is
130 achieved when more than 200 images are used in the ensemble-average calculation [5]). In this case, each velocity vector
131 is a relative velocity (\mathbf{W}) which, by definition, is mathematically equivalent to subtracting the tangential term ($\mathbf{\Omega} \times \mathbf{r}$) due
132 to the impeller motion from the absolute velocity (\mathbf{U}) in the velocity triangle [33]. Thus, the procedure of removing the
133 angular displacement of the impeller is equivalent to calculating the term $\mathbf{U}_i - \mathbf{\Omega} \times \mathbf{r}_i$ in the following equation:

$$\langle \mathbf{W} \rangle = \frac{1}{n} \sum_{i=1}^n \mathbf{W}_i = \frac{1}{n} \sum_{i=1}^n (\mathbf{U}_i - \mathbf{\Omega} \times \mathbf{r}_i) \quad (1)$$

134 where n is the number of images that compose the ensemble-averaged velocity, $\mathbf{\Omega}$ is the angular velocity, which is
135 proportional to the rotational speed, and \mathbf{r} is the radial position, which varies from the inner to the outer radius in the

136 impeller. Working with relative velocity vectors is convenient especially for regions inside or close to the impeller. The
 137 magnitude of this relative velocity can be normalized by the tangential velocity calculated at the tip of the blades, at the
 138 impeller outlet, where $r = r_2$. In this case, $W^* = W/(\Omega r_2)$.

139 Figure 5 illustrates the procedure that transforms raw images from a PIV test into an average field of relative velocity.

140 For a position (x_i, y_i) in the impeller and an operating condition (N, Q_w) of the pump, the comparison of two different
 141 vectors is an interesting strategy to evaluate the influence of the axial position (z) on the flow dynamics. Therefore, a
 142 vector found on the periphery (e.g. \mathbf{W}_{shroud}) can be compared with a correspondent vector found on the central plane
 143 (\mathbf{W}_{center}), through a cross product that offers an estimation of the alignment between both vectors, measured by angle ϕ :

$$\phi = \arcsin \left(\frac{\langle \mathbf{W}_{shroud} \rangle}{|\langle \mathbf{W}_{shroud} \rangle|} \times \frac{\langle \mathbf{W}_{center} \rangle}{|\langle \mathbf{W}_{center} \rangle|} \right) \quad (2)$$

144 The velocity vectors obtained from the PIV tests have components only in the x and y directions, without any values in
 145 z . It is thus possible to write $\mathbf{W} = \mathbf{w}_x + \mathbf{w}_y$. Besides, each instantaneous velocity vector can be understood as the sum of
 146 an average velocity with a fluctuation in time. From this definition, an analysis of the turbulence levels can be performed
 147 in the impeller. Since only two components of the velocity fluctuations are available in the 2D2C-PIV measurements, the
 148 turbulent kinetic energy, K_{2D} , is calculated as [14]:

$$K_{2D} = \frac{1}{2} \left(\langle \mathbf{w}'_x \rangle^2 + \langle \mathbf{w}'_y \rangle^2 \right) \quad (3)$$

149 where $\langle \mathbf{w}'_x \rangle^2$ and $\langle \mathbf{w}'_y \rangle^2$ are variances of the velocity fluctuations in time, calculated as: $\langle \mathbf{w}'_x \rangle^2 = \frac{1}{n} \sum_{i=1}^n (\mathbf{w}_{x,i} - \langle \mathbf{w}_x \rangle)^2$
 150 and $\langle \mathbf{w}'_y \rangle^2 = \frac{1}{n} \sum_{i=1}^n (\mathbf{w}_{y,i} - \langle \mathbf{w}_y \rangle)^2$. The resultant turbulent kinetic energy can be normalized by the linear velocity of
 151 the blade tip, $K_{2D}^* = K_{2D}/(\Omega r_2)^2$ [14], where r_2 is the outer radius or half the outer diameter d_2 . To improve the quality
 152 of instantaneous vectors, we use pre- and post-processing procedures (applying filters, removing invalid vectors, etc).
 153 Furthermore, the adaptive PID method is configured to only accept vectors that meet certain criteria (peak height, height
 154 ratio, signal-to-noise ratio) and replace, in the last iteration, those that do not meet the minimum requirements.

155 3 RESULTS AND DISCUSSIONS

156 This section presents the main results obtained from acquiring and processing PIV images on different axial planes of
 157 the transparent pump. Velocity fields (subsection 3.1) and profiles (subsection 3.2) in different positions and conditions,
 158 as well as turbulence (subsection 3.3), compose the results discussed here.

159 **3.1 Velocity fields and deviations between vectors.** The fields of ensemble-averaged relative velocities measured
 160 on the central plane of the impeller, \mathbf{W}_{center} (Eq. 1), are presented in Fig. 6 for the six conditions investigated in this
 161 manuscript. The impeller rotates in the clockwise direction.

162 The relative velocity is very low when the pump operates at low flow rates. In the condition close to shut-off, Figs. 6(a)
 163 and 6(b) reveal that the flow topology is characterized by the presence of vortices and reverse flow. This is valid mainly

164 at the left side of the impeller, which is close to the volute tongue and adjacent to the first half of the volute spiral, where
165 r_{vol} is relatively small. Then, at the design point, Figs. 6(c) and 6(d) show that the flow becomes more well-organized and
166 aligned with the curvature of the blades, as expected for the BEP. In the condition corresponding to open-flow, however,
167 Figs. 6(e) and 6(f) indicate that the vectors undergo a deviation toward the suction surface of the blades. The structure
168 becomes similar to half a counterclockwise vortex as the boundary layers tend to detach from the pressure blades. These
169 observations agree with results presented in Perissinotto et al. [5] for other tests focused on the central plane of the impeller.

170 The magnitude of the relative velocity vectors in Fig. 6 is limited to the range $1.2 < W_{center} < 3.2$ m/s. The velocity at
171 the tip of each blade is $\Omega r_2 = 3.5$ m/s for $N = 600$ rpm and $\Omega r_2 = 5.2$ m/s for $N = 900$ rpm, where $r_2 = d_2/2$ is the outer
172 radius of the impeller. It means that the scale used in W_{center} represents 23% to 69% of the velocity Ωr_2 , depending on
173 the condition.

174 Other velocity fields are presented in Fig. 7, for three flow rates at $N = 600$ rpm. In this case, the velocity is measured on
175 the peripheral laser planes that illuminate the flow next to the hub (\mathbf{W}_{hub}) and shroud (\mathbf{W}_{shroud}). From a visual inspection
176 of the images, we can observe that there are no major differences between the central and peripheral flow fields. However,
177 divergences exist, and become more evident when streamlines are analyzed together with the velocities.

178 Thus, Fig. 8 compares the streamlines obtained on the three planes, for each flow rate Q at the rotational speed $N = 600$
179 rpm. These streamlines are superimposed on contour plots of the relative velocity, W , normalized by the tangential velocity
180 due to the impeller rotation, Ωr , calculated on the tip of the blades at the impeller exit, r_2 . Therefore, the plots contain
181 values of $W^* = W/(\Omega r_2)$, as explained in subsection 2.3.

182 An evaluation of the alignment between vectors is another way of highlighting the differences between these velocity
183 fields. The idea is to calculate the angle ϕ (Eq. 2) between the peripheral and central velocity vectors in each position of
184 the two-dimensional field. The contour plots from Fig. 9 highlight the regions where the vectors are most misaligned. At
185 shut-off conditions, deviations occur due to variations in the position of the vortices, as depicted in Figs. 9(a) and 9(b), in
186 which $-90^\circ < \phi < 90^\circ$. At the BEP, the most relevant deviations are found in the upper left channels, where the angles
187 are limited to the scale $-60^\circ < \phi < 60^\circ$, as shown in Figs. 9(c) and 9(d). Then, at the open-flow condition, the changes in
188 the vectors are concentrated mainly on the pressure blades, as depicted in Figs. 9(e) and 9(f), with deviations in the range
189 $-60^\circ < \phi < 60^\circ$.

190 The analysis was extended for $N = 900$ rpm in Fig. 10 and we note that the results are similar to those obtained at
191 $N = 600$ rpm. The main differences include the positions of vortex structures at the lowest water flow rate. In addition,
192 at the BEP, the blue regions where $\phi > 60^\circ$ in Fig. 9(c) disappear when N increases, as can be seen in Fig. 10(c). Such
193 characteristics are probably related to the dynamics of the water flow. However, we analyzed the raw PIV images and
194 observed that: 1) For flow rates close to the shut-off, part of the tracer particles tend to segregate and deposit in the lower
195 regions of the pump, so that there is a lack of particles on the images; 2) At the BEP, some particles accumulate in the
196 vicinity of the blades, giving rise to white spots in the images. These situations have possibly impaired the quality of the
197 image processing, despite our attempts to minimize the issues by applying pre- and pos-processing techniques. Yet, it is
198 worth mentioning that the clearance leakage becomes more severe as the rotational speed increases, so that the flow rate

199 measured by the flow meter becomes different from the actual flow rate in the impeller. This fact may cause some influence
200 in the fields obtained at different N .

201 For both N , the deviations between vectors are more significant near the hub than near the shroud. We should remember
202 that the shroud is made of polymeric material (acrylic), while the hub is made of metallic material (aluminium) with a layer
203 of paint. These parts have different surface roughness, which may affect the flow behavior. In addition, another important
204 feature that influences the flow is the geometry of the pump prototype regarding the assembly of the closed radial impeller
205 inside the vaneless volute diffuser. The hub faces the rear part of the pump stage, while the shroud is spaced 2 mm away
206 from the front part, due to the presence of a gap downstream from the impeller.

207 This detail of the visualization section is shown in Fig. 11. The flow encounters an “opening” as it leaves the impeller
208 next to the internal surface of the shroud. The fluid is free to move towards the front cover of the volute, configuring a
209 three-dimensional mean flow, with a velocity term in the z direction. On the other hand, when exiting close to the hub
210 surface, the flow continues adjacent to the solid walls. The impeller hub is mounted at the same height as the volute
211 bulkhead, that is, these parts have identical dimensions in the z direction.

212 Differences between the flow in the shroud and the flow in the hub are also a consequence of the geometry of components
213 placed upstream from the impeller. They possibly cause the flow to change its direction as it enters the channels. This
214 effect will be discussed in subsection 3.2.

215 **3.2 Velocity profiles.** The current subsection aims to discuss the differences between the velocity profiles obtained on
216 the three axial planes of the impeller. These profiles are thus evaluated at three radial positions $r = [27, 38, 50]$ mm, which
217 correspond to $r^* = r/r_2 = [0.5, 0.7, 0.9]$ when normalized by the outer radius of the impeller. The analysis is focused on
218 two channels: CA is far from the volute tongue, while CB is very close to it. The exact location of CA and CB can be
219 defined as follows: CA is the only channel whose entrance is completely within the first quadrant of the impeller; CB is
220 the only channel whose exit is completely within the third quadrant of the impeller.

221 Each channel is defined by a circumferential position, c , which can be normalized. In this case, $c^* = 0.0$ refers to the
222 suction surface of the blade (SB) and $c^* = 1.0$ corresponds to the pressure surface (PB), where $c^* = (c - c_{SB}) / (c_{PB} - c_{SB})$.
223 Examples of velocity profiles determined at the central plane are available in Fig. 12 for the shut-off, BEP, and open-flow
224 conditions at $N = 600$ rpm. The figure also highlights the three radial positions r^* , the circumferential direction c , and the
225 location of channels CA and CB.

226 Figures 13, 14 and 15 compare the profiles acquired on the central plane (according to Fig. 12) with the ones measured
227 on the peripheral planes (i.e., next to hub or shroud), for $N = 600$ rpm, at flow rates associated with the shut-off, BEP, and
228 open-flow conditions, respectively. The y-axis of the graphs contains the magnitude of the relative velocity normalized by
229 the tangential velocity at the impeller exit, $W^* = W / (\Omega r_2)$, as explained in subsection 2.3.

230 As a general rule, the ensemble-averaged relative velocity increases as the flow rate increases. However, for a constant
231 flow rate, this velocity tends to decrease as the radial position changes from the inner ($r_1^* = 0.5$) to the outer region
232 ($r_3^* = 0.9$). This fact is related to the geometry of the channel, which is similar to a divergent nozzle – in other words, as
233 the radial position increases, the cross-sectional area of the channel also increases, so that the velocity decreases for a fixed

234 flow rate. Furthermore, the velocity tends to have higher magnitudes in channel CA than in channel CB. This behavior is a
235 consequence of the interaction between the impeller and volute, as the solid walls that compose the volute spiral influence
236 the flow in the impeller channels [5].

237 From Figs. 13, 14 and 15, we observe that the relative velocity on the peripheral planes is similar to that on the central
238 plane. In most conditions, the curves for W and W^* present the same shape, independently of the axial position of the
239 plane illuminated during the PIV tests. For the lowest flow rate, differences are detected in channel CB (Figs. 13(b), 13(d),
240 13(f)). The curves appear to be translated relatively to each other, as if there was a phase shift between the peaks and
241 valleys. This result is possibly associated with the position of vortex structures, which vary slightly as a function of the
242 plane considered (see Fig. 8(a)).

243 At the BEP, the curves for channel CB (Figs. 14(b), 14(d), 14(f)) reveal that the velocity has a lower intensity on the
244 hub plane than on the other planes. We observe the same trend in channels CA and CB at the highest Q_w (Fig. 15). In fact,
245 such differences are probably an effect of the surface finish, as the hub has a larger roughness than the shroud, because the
246 former is a machined aluminum part whereas the latter is a polished acrylic plate. There is possibly a relationship between
247 the flow fields in the impeller and the material used in its manufacture. This fact could be further explored by the industry
248 in the search for more efficient impellers.

249 Nevertheless, at the highest flow rate (Fig. 15) most velocities measured on the shroud plane (close to the walls) are
250 higher than the ones measured on the central plane (far from the walls). We should remember that the flow in the impeller
251 is also influenced by the geometry of the other components mounted in the centrifugal pump. Before entering the impeller,
252 the fluid undergoes a change in its direction, which causes a deviation in the streamlines towards the internal surface of the
253 shroud, especially at high flow rates. As a result, the velocity profiles become more distorted.

254 This is illustrated in Fig. 16. In the drawing, the yellow component is a diffuser adapted from the ESP P23 model. Its
255 purpose is to guide the fluid into the impeller, reducing the occurrence of collisions with the solid walls [32]. However,
256 although the part smooths out the change in the flow direction, it still happens, on a 180° curve.

257 The profiles drawn in Fig. 16 were inspired on an analogy between pumps and ducts. Inside a pipeline, a considerable
258 length is required after a bend for the flow to recover its full development. In fact, the flow in a tight bend is characterized
259 by distorted velocity profiles with the occurrence of different secondary flow patterns depending on the transverse and
260 longitudinal locations [36–38]. An even worse condition is expected in the impeller, as the geometry is complex (curved
261 channel with a variable cross-sectional area) and the distances are short (just a few millimeters from inner to outer radius).
262 This would be another opportunity for improvements in the design of commercial pumps: the components located upstream
263 from the impeller may be modified to beneficially influence the flow behavior and the efficiency of energy transfer in the
264 impeller.

265 We extended the analysis for $N = 900$ rpm and obtained similar results. The curves of relative velocity present similar
266 shapes for both rotational speeds, but the magnitudes increase as N rises. Thus, the observations made for $N = 600$ rpm
267 are also valid for $N = 900$ rpm, and other graphs will be suppressed because they do not provide new information to the
268 discussions.

269 **3.3 Turbulent kinetic energy.** The turbulent kinetic energy, K_{2D} , is a measure of the energy associated with velocity
 270 fluctuations in time. It is thus calculated from the variances $\langle \mathbf{w}'_x \rangle^2$ and $\langle \mathbf{w}'_y \rangle^2$ in the x and y directions (Eq. 3), obtained by
 271 subtracting the average from the instantaneous velocity vectors in each interrogation window of the PIV fields. Figure 17
 272 shows the turbulent kinetic energy computed in the pump stage at $N = 600$ rpm, for three flow rates and three axial
 273 positions of the laser planes. The values are normalized by the linear velocity of the blade tip. Hence, $K_{2D}^* = K_{2D}/(\Omega r_2)^2$,
 274 in accordance with subsection 2.3.

275 In Fig. 17, the turbulent kinetic energy evaluated on the central plane agrees well with results from Perissinotto et al.
 276 [5]. The K_{2D}^* levels are the lowest when the pump works under the best efficiency point (Fig. 17(b)), but they become
 277 much more intense when the device operates at off-design conditions (Figs. 17(a) and 17(c)). Besides, Fig. 17 reveals that
 278 the turbulent kinetic energy is higher on the peripheral planes than on the central plane. This is an effect of the interaction
 279 between the fluid and solid walls, characterized by significant velocity gradients and shear stresses, which increase the
 280 turbulence production and thus make K_{2D} more intense. As we can observe, the contour plots related to the hub and
 281 shroud present large areas covered by red spots, revealing large regions where K_{2D}^* is higher than the limits of the scales.
 282 Furthermore, it is important to remember that the flow is affected by the parts placed upstream the impeller, as explained in
 283 the subsection above (see Fig. 16). In addition to influencing the average velocity profile, it is possible that the component
 284 installed behind the impeller also modifies the velocity fluctuations, applied in the K_{2D} calculation.

285 Another interesting result is the turbulence level within the volute spiral. By comparing the contour plots, we observe
 286 that K_{2D}^* in the volute is higher on the hub and lower on the shroud plane. This fact is related to the geometry of the pump
 287 stage. As explained in subsection 3.1, Fig. 11, there is a gap (2 mm) between the external surface of the shroud and front
 288 cover of the pump stage. As a consequence, the laser plane that illuminates the internal surface of the shroud is almost 5
 289 mm away from the solid walls of the volute. We can thus expect that the K_{2D}^* values are relatively lower in this case, as
 290 the velocity gradient and shear stress tend to decrease in locations far from walls.

291 As stated in the last paragraphs, the turbulent kinetic energy is evaluated from velocity fluctuations measured in each
 292 interrogation window of the PIV images. Hence, the sum of each K_{2D} value, in each one of the n_{IW} windows, express an
 293 estimation of the total energy associated with turbulence in the entire impeller:

$$K_{2Dsum} = \sum_{i=1}^{n_{IW}} K_{2D} \quad (4)$$

294 An analysis of the total turbulent kinetic energy was conducted and the results are presented in Fig. 18, for $N = 600$ rpm.
 295 The y -axis contains K_{2Dsum}^* , which is normalized by K_{2Dsum} on the central plane, at the BEP. As this condition (BEP on
 296 central plane) has the lowest turbulent levels, it is considered here as the reference for comparisons, corresponding to the
 297 basic unity of normalized values.

298 From Fig. 18, we can observe that the total turbulent kinetic energy computed on the peripheral planes reaches values
 299 eight times greater than K_{2Dsum}^* measured on the central plane. For a more complete analysis, additional experiments
 300 were performed at flow rates corresponding to 30%, 80%, and 120% of the BEP. The data plotted on the graph is adjusted

301 by polynomial fits, following the procedure firstly proposed in Perissinotto et al. [5]. The lowest points of these curves
302 occur around the BEP, while the highest points are found at the conditions furthest from the BEP, i.e., shut-off and open-flow.

303

304 The investigations were repeated for $N = 900$ rpm. We could note that the results are roughly the same as those for
305 $N = 600$ rpm. Hence, the observations made for the lowest rotational speed are also valid for the highest one. Results for
306 $N = 900$ rpm will be thus suppressed as they do not include any new information to the discussions developed here.

307 **3.4 Database for future studies.** The experimental data acquired in the tests is available at [https://gitlab.com/](https://gitlab.com/perissinotto/PIV_pump_planes)
308 [perissinotto/PIV_pump_planes](https://gitlab.com/perissinotto/PIV_pump_planes) for download. The database contains quantities such as average velocities, velocity fluc-
309 tuations, and turbulent kinetic energy as a function of the x, y position of the centrifugal pump, measured at different
310 rotational speeds and flow rates, according to the test matrix of this manuscript. Thus, the results for $N = 900$ rpm (that
311 were suppressed in the last subsections) are included in the folder as well.

312 Furthermore, the path https://gitlab.com/perissinotto/PIV_pump_planes contains the 3D technical drawings created dur-
313 ing the design of the transparent centrifugal pump. These files were originally elaborated on the CAD software *SolidEdge*,
314 academic version, by Siemens. Other two file extensions (IGS and STP) are also provided in the repository for the user's
315 convenience.

316 Our intention with sharing these files is to allow other researchers to validate the accuracy and fidelity of their models
317 against our experimental data. In this context, the technical drawings of our pump can be used to generate computational
318 meshes and carry out numerical simulations. This may lead to future collaborative studies between diverse research groups.

319 4 CONCLUSIONS

320 In this manuscript, experiments using particle image velocimetry (PIV) were carried out to study the single-phase water
321 flow in the closed radial impeller and vaneless volute diffuser of a centrifugal pump. The processed images provided
322 information on the velocity fields, velocity profiles, and turbulence levels in three different axial planes of the impeller: i)
323 a central plane equidistant from the front and rear covers; ii) a peripheral plane adjacent to the internal surface of the front
324 cover (shroud); iii) a peripheral plane adjacent to the internal surface of the rear cover (hub). After analyzing the results
325 and comparing them with literature [5], we can conclude that:

326 1. The ensemble-averaged relative velocity in the impeller is very dependent on the flow rate (Q). At low Q , close to
327 the shut-off, the flow topology is characterized by the presence of vortex structures that partially block the channels. At the
328 best efficiency point (BEP), the flow is well-organized, with streamlines following the blade curvature. At high Q , close
329 to the open-flow, the streamlines are diverted towards the suction side of the blades, so that the fluid path becomes longer
330 and detached from the pressure blades. These results are valid for the two rotational speeds (N) and three axial planes
331 investigated here. This close connection between flow field and flow rate in the impeller is a consensus among current
332 researchers, so that our observations completely agree with results from other studies available in literature.

333 2. Significant differences are observed when we compare the velocity vectors on the central plane with those on the
334 peripheral planes. These divergences are identified as angular deviations between two vectors placed at the same x, y point

335 on two different planes. The misalignment is more relevant on the hub plane due to geometrical attributes of the centrifugal
336 pump: i) a higher roughness of the hub in comparison with the shroud; ii) the existence of a gap between the shroud and
337 the front cover of the volute, which may influence the flow within the impeller. Under the shut-off, the deviations are even
338 more evident, because there are slight variations in the position of vortex structures depending on the plane investigated.
339 However, the deviations under BEP and open-flow conditions do not have a well-defined behavior: there are regions on the
340 peripheral plane in which the vectors have positive (clockwise) deviations in relation to vectors in the central plane; but
341 there are other regions in which this deviation is negative (counterclockwise). Such interesting variations can be confirmed
342 by analyzing the streamlines and velocity profiles.

343 3. As a consequence of the above conclusions, the velocity profiles are dependent on Q and also on the radial (r) and
344 circumferential (c) positions. Regarding the magnitude of the ensemble-averaged relative velocity, it is possible to observe
345 that this velocity is generally lower on the hub plane and higher on the shroud plane. This is an effect of: i) once again,
346 the surface finish, with a higher roughness on the hub than on the shroud; ii) the influence of other components mounted
347 upstream from the impeller, which smoothly guide the flow, but may favor the deflection of the fluid towards the internal
348 surface of the shroud. The same findings are valid for both N . However, different results may be obtained if the geometry
349 of the pump is modified and its internal components replaced, and this could be an opportunity for future studies.

350 4. On the central plane, as expected, the turbulent kinetic energy measured at the shut-off and open-flow presents higher
351 values than that one evaluated at BEP, this result being well-known to scholars. The turbulence levels are more relevant: i)
352 at the impeller-volute boundary, where there are significant velocity differences, for shut-off; ii) at the tip of the blades, due
353 to wakes formed at the trailing edges, for BEP; iii) on the pressure side of the blades, where boundary layer detachment
354 occurs, for open-flow. In addition, the turbulent kinetic energy is higher on the peripheral planes, as the presence of walls
355 tends to make the intensity of velocity fluctuations increase. A measure of the total turbulence across the impeller provides
356 values eight times greater on the periphery (shroud plane, open-flow) when compared to the center (mid-axial plane, BEP).

357 The observations made in this study contribute to a better understanding of the flow subject to the rotating conditions
358 found in a pump impeller. The findings indicate possible points of improvement in the design of centrifugal pumps, such as
359 the reduction of surface roughness and the addition of elements to homogenize the flow. From the availability of the PIV
360 database and technical drawings of the transparent pump, it is expected that new studies can continue this investigation,
361 with the proposition of phenomenological models or execution and validation of numerical simulations, for example.

362 Suggestions for future studies include: i) Carry out new experiments using time-resolved PIV to further explore the
363 relationship between the characteristics of velocity distribution in different planes and the complex flow phenomena
364 occurring in pump impellers, such as rotational stalls; ii) Expand the investigations for fluids more viscous than water,
365 such as glycerol or mineral oil, to address the effects of viscosity on the flow behavior; iii) Try new impeller designs
366 to investigate the flow characteristics as a function of impeller diameter, blade curvature, number of channels, among others.

367

368 **Nomenclature / List of Symbols**

Symbol	Description
Q_w	Water flow rate
Q_{BEP}	Water flow rate at BEP
ρ	Water density
T	Temperature at pump intake
P	Pressure at pump intake
ΔP	Pressure increment produced by pump
N	Rotational speed of pump impeller
Ω	Angular speed of pump impeller
H	Pump head
d_1, d_2	Impeller inner and outer diameters
r_1, r_2	Impeller inner and outer radii
h_{imp}	Impeller height
r_{vol}	Radius of volute spiral
h_{vol}	Height of volute spiral
\mathbf{U}	Absolute velocity vector
\mathbf{W}	Relative velocity vector
$\langle \mathbf{W} \rangle$	Average relative velocity vector
W	Relative velocity
W^*	Normalized average relative velocity
$\langle \mathbf{w}_x^2 \rangle^2, \langle \mathbf{w}_y^2 \rangle^2$	Variances of velocity fluctuations
ϕ	Angle of deviation between velocity vectors
K_{2D}	Turbulent kinetic energy
K_{2D}^*	Normalized turbulent kinetic energy
K_{2Dsum}	Sum of K_{2D} values throughout the contour plot
K_{2Dsum}^*	Normalized K_{2Dsum}
x, y	Position in Cartesian coordinates
r, c	Position in polar coordinates
r^*, c^*	Normalized radial and circumferential positions
n	Number of images in the dataset
n_{IW}	Number of interrogation windows in the image

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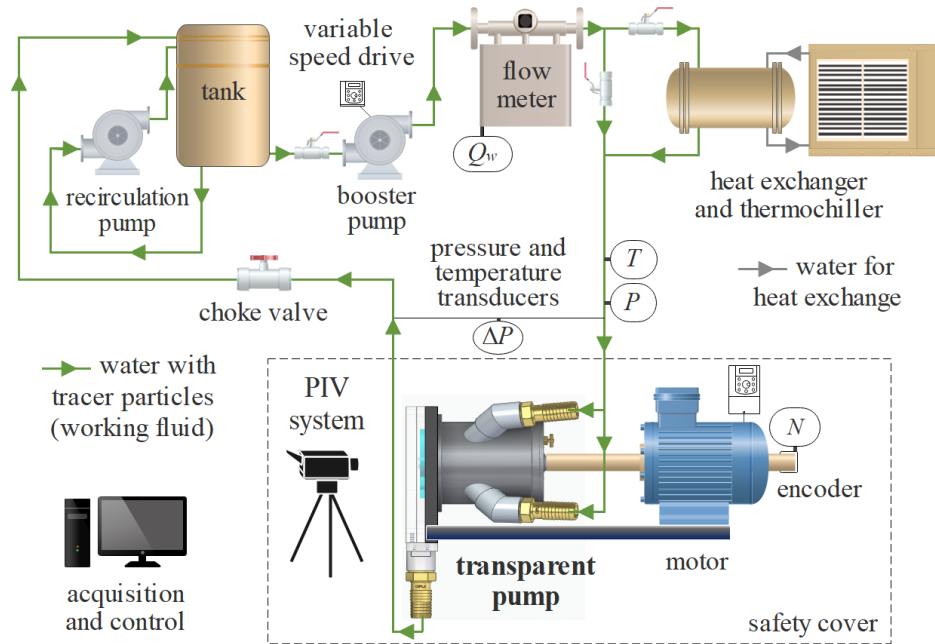


Fig. 1 Layout showing the main components of the test facility.

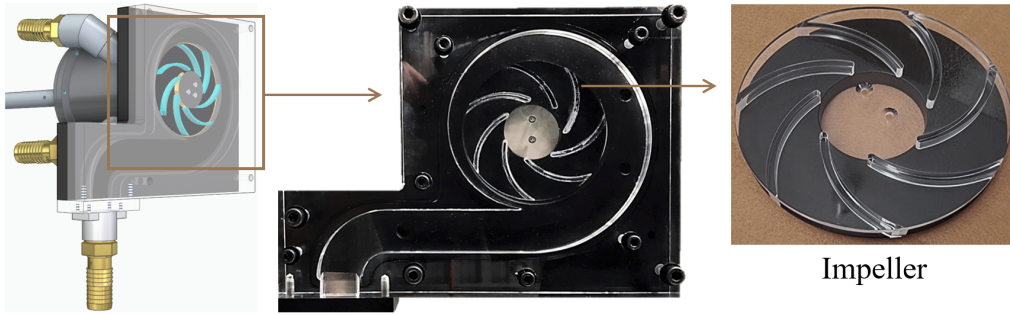


Fig. 2 Visualization section of the transparent pump with impeller and volute.

Table 1 Test matrix with six operating conditions.

N	Q_w - absolute and relative to Q_{BEP}		
	$\leq 10\%$	$\sim 100\%$ (BEP)	$\sim 160\%$
600 rpm	0.15 m ³ /h	1.5 m ³ /h	2.4 m ³ /h
900 rpm	0.15 m ³ /h	2.2 m ³ /h	3.6 m ³ /h

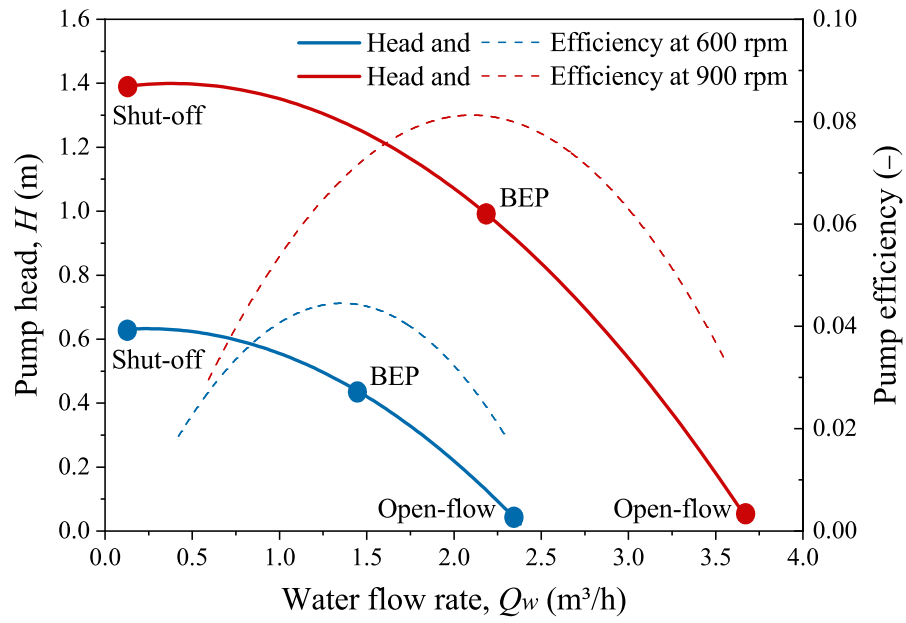


Fig. 3 Head and efficiency curves of the transparent pump operating with water.

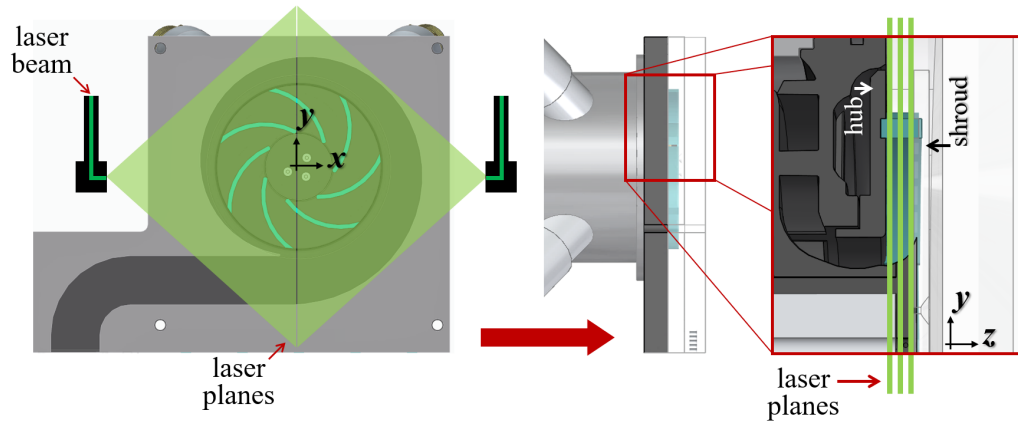


Fig. 4 Configuration of the two laser planes in three axial positions (z direction).

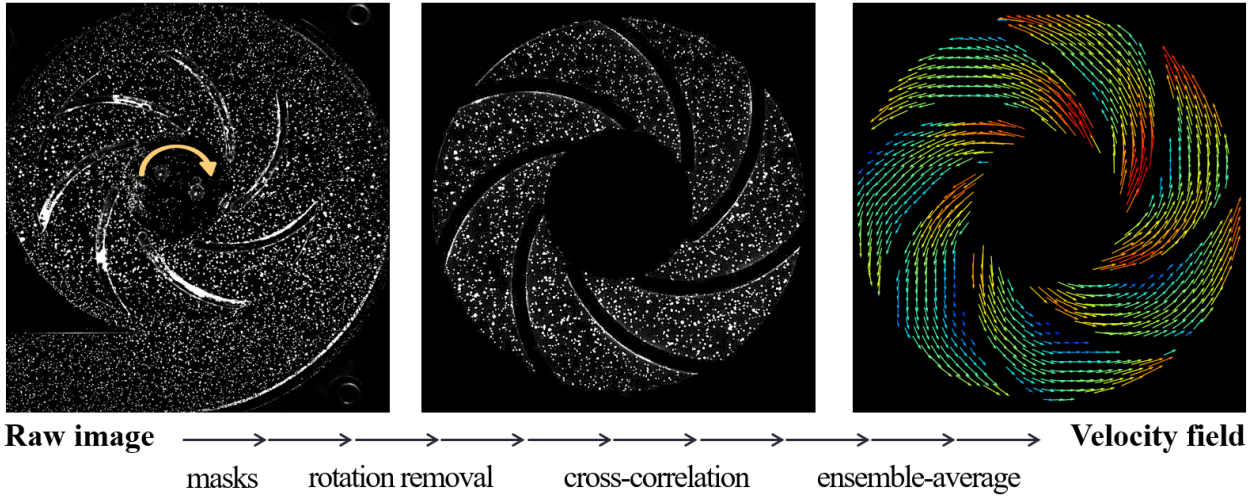


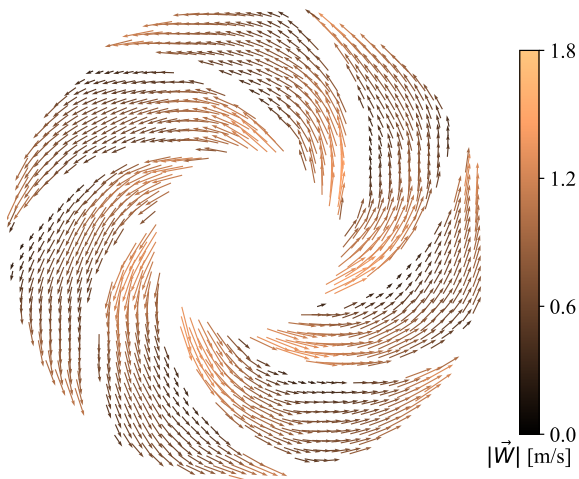
Fig. 5 Illustration of image processing: from raw images to a velocity field.



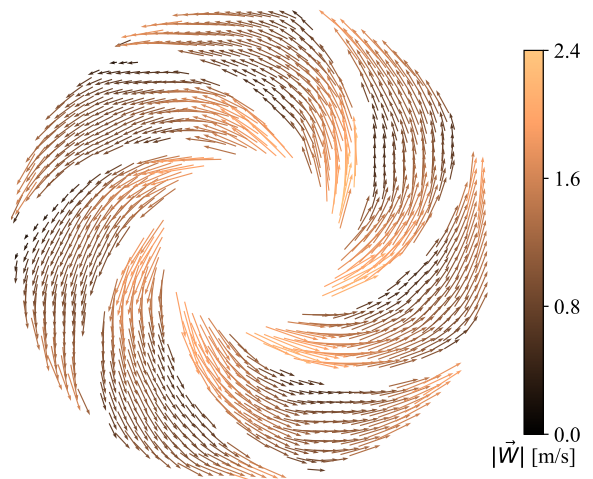
(a) $Q_w = 0.1 Q_{BEP}$ at $N = 600$ rpm



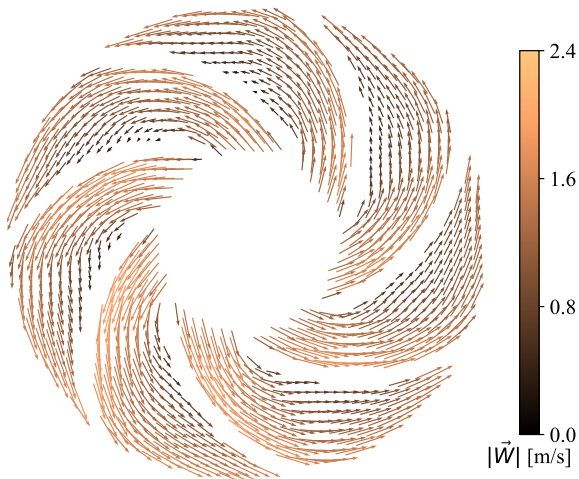
(b) $Q_w = 0.1 Q_{BEP}$ at $N = 900$ rpm



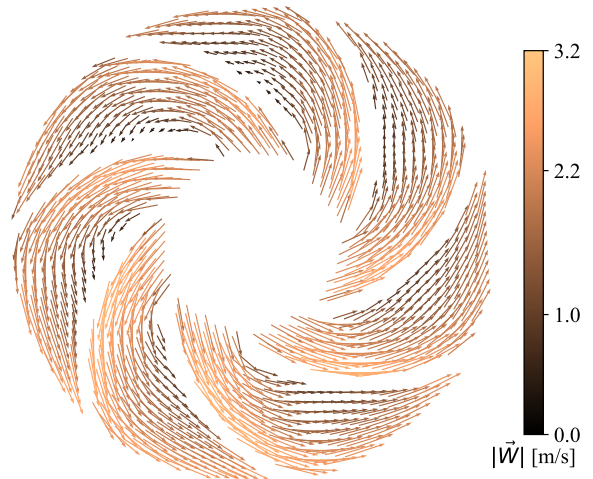
(c) $Q_w = 1.0 Q_{BEP}$ at $N = 600$ rpm



(d) $Q_w = 1.0 Q_{BEP}$ at $N = 900$ rpm



(e) $Q_w = 1.6 Q_{BEP}$ at $N = 600$ rpm



(f) $Q_w = 1.6 Q_{BEP}$ at $N = 900$ rpm

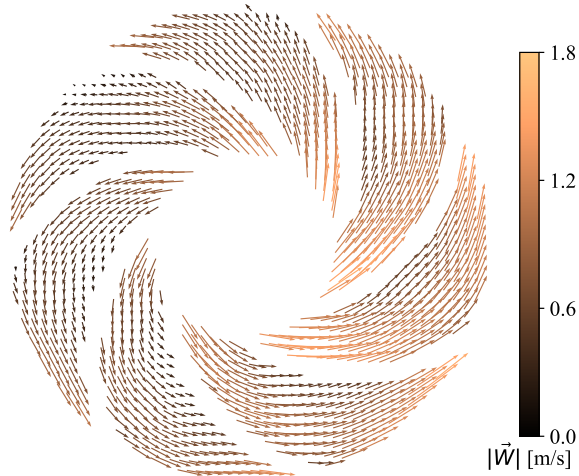
Fig. 6 Average relative velocity W_{center} measured on the central plane of the impeller.



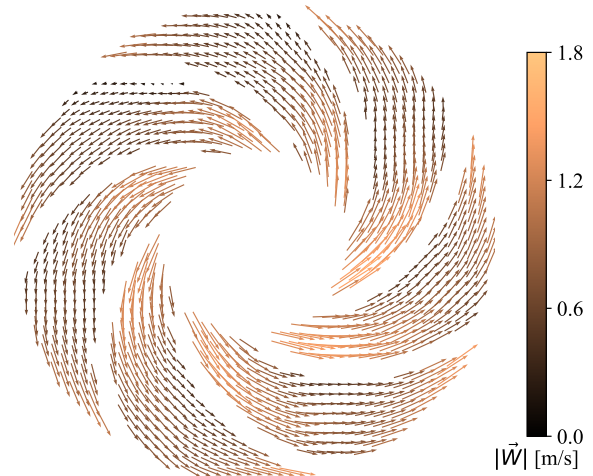
(a) $Q_w = 0.1 Q_{BEP}$ measured next to hub plane



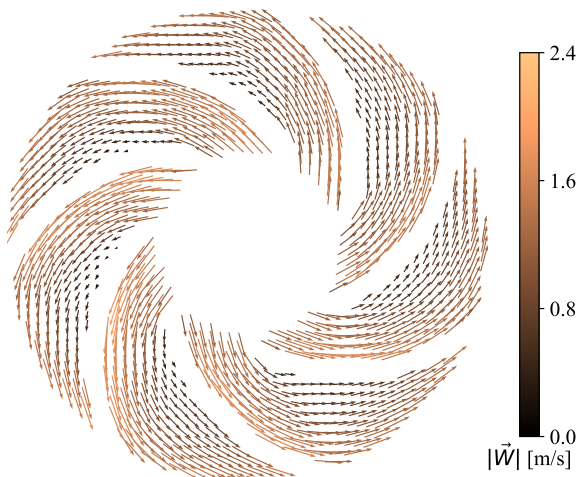
(b) $Q_w = 0.1 Q_{BEP}$ measured next to shroud plane



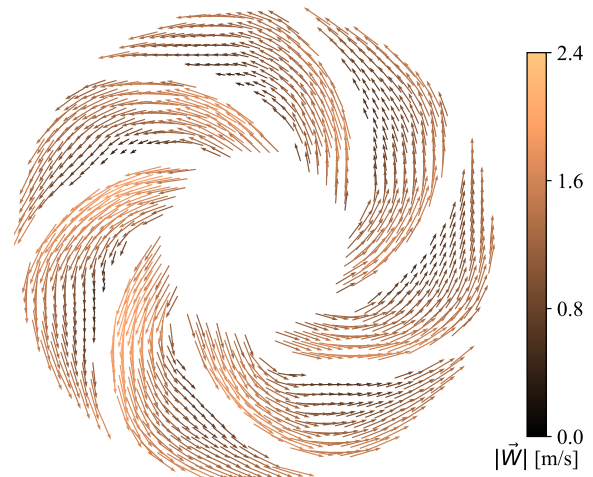
(c) $Q_w = 1.0 Q_{BEP}$ measured next to hub plane



(d) $Q_w = 1.0 Q_{BEP}$ measured next to shroud plane

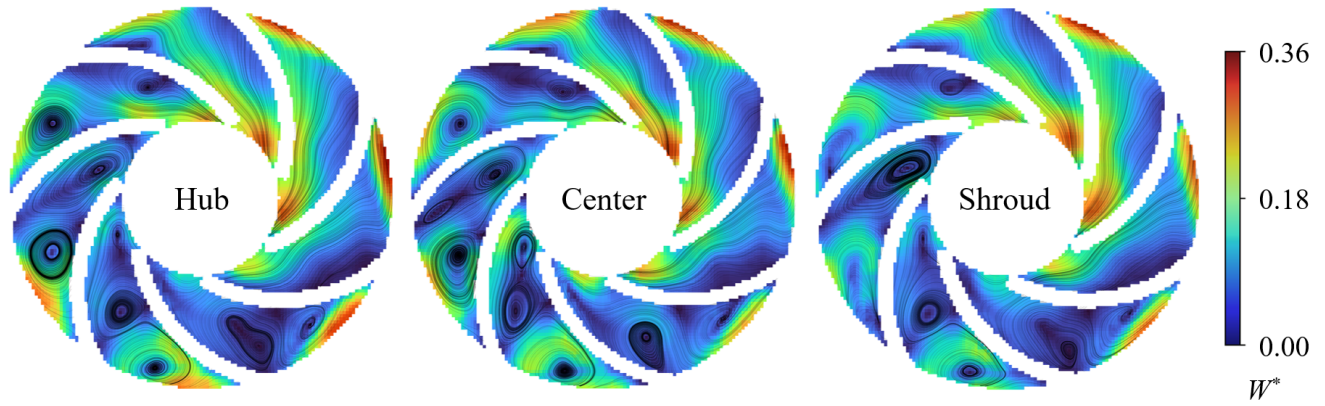


(e) $Q_w = 1.6 Q_{BEP}$ measured next to hub plane

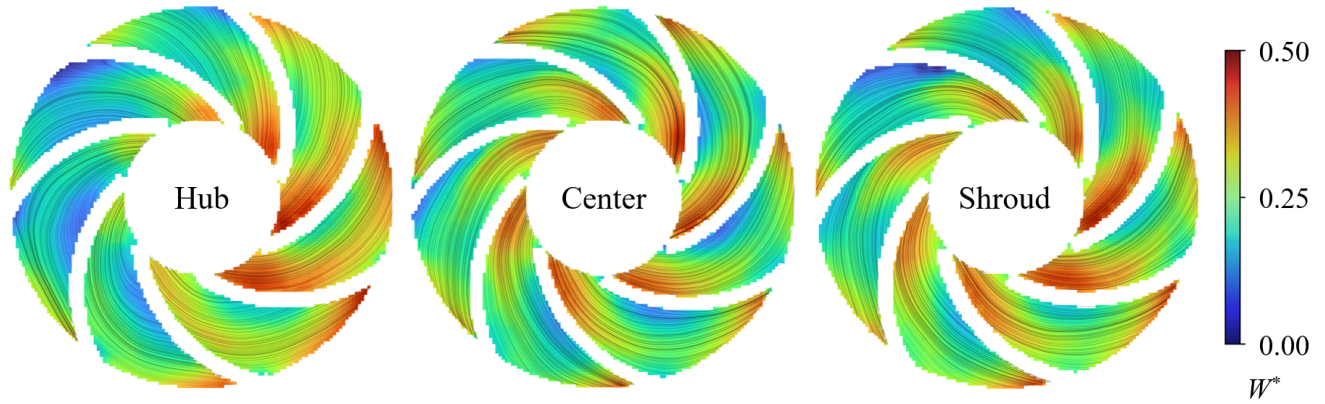


(f) $Q_w = 1.6 Q_{BEP}$ measured next to shroud plane

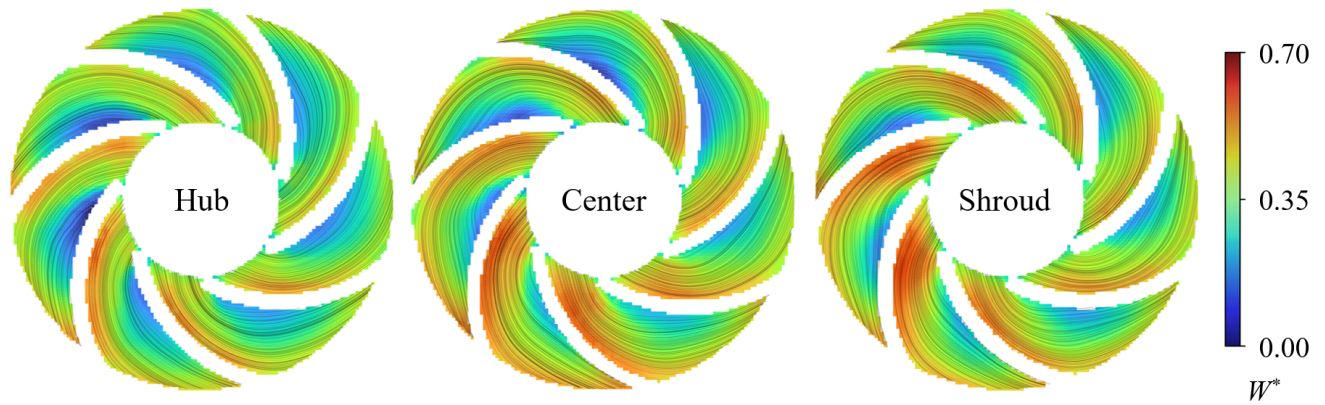
Fig. 7 Average relative velocity W_{hub} and W_{shroud} , measured on peripheral planes, at $N = 600$ rpm.



(a) $Q_w = 0.1 Q_{BEP}$, measurements performed on three axial planes

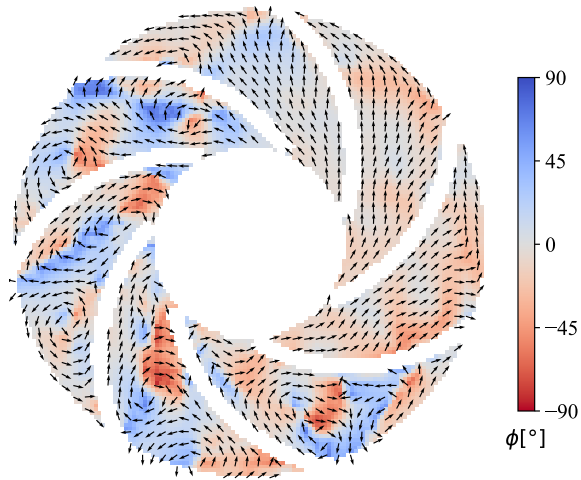


(b) $Q_w = 1.0 Q_{BEP}$, measurements performed on three axial planes

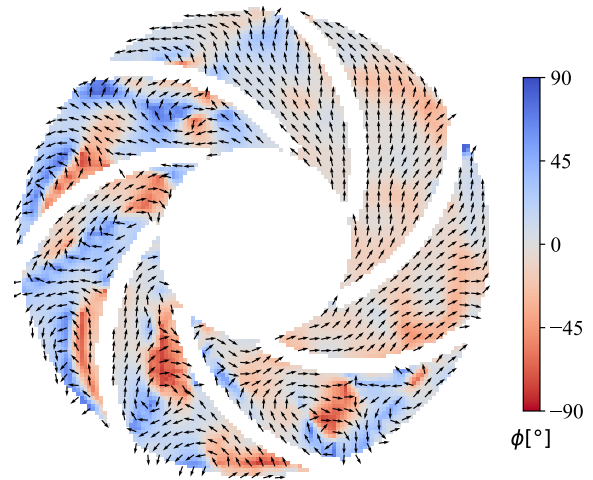


(c) $Q_w = 1.6 Q_{BEP}$, measurements performed on three axial planes

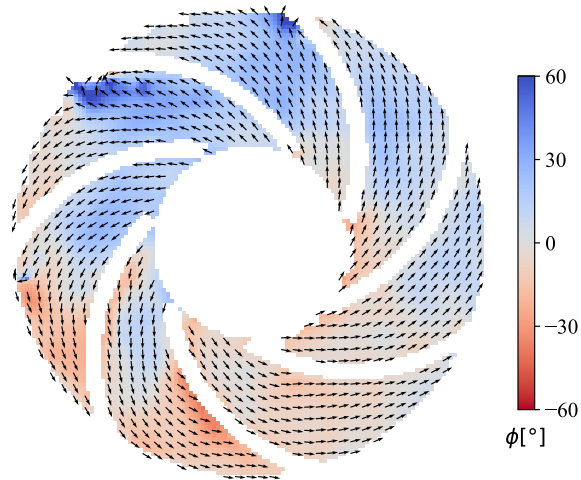
Fig. 8 Streamlines superimposed on contour plots of normalized relative velocity, $W^* = W/(\Omega r_2)$, measured on the central and peripheral planes in the impeller and volute. $N = 600$ rpm.



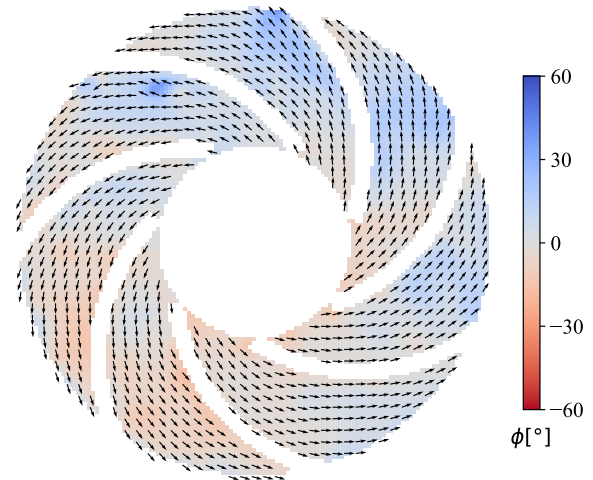
(a) $Q_w = 0.1 Q_{BEP}$ measured next to hub plane



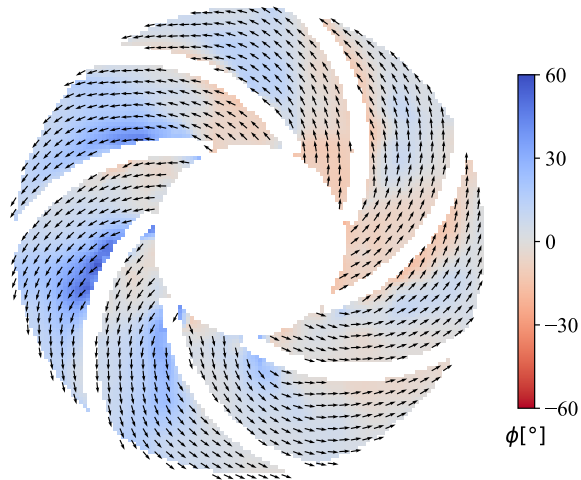
(b) $Q_w = 0.1 Q_{BEP}$ measured next to shroud plane



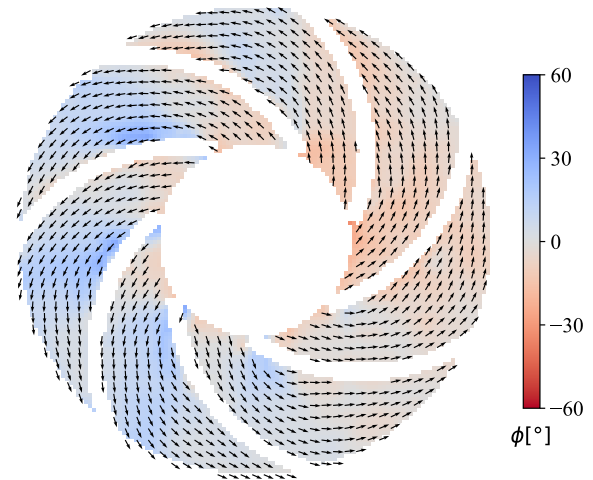
(c) $Q_w = 1.0 Q_{BEP}$ measured next to hub plane



(d) $Q_w = 1.0 Q_{BEP}$ measured next to shroud plane

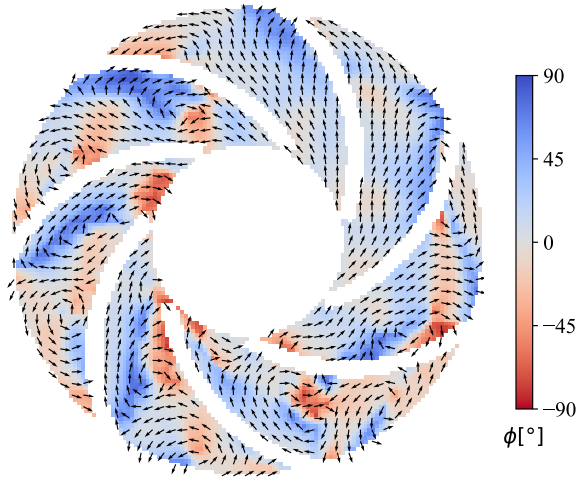


(e) $Q_w = 1.6 Q_{BEP}$ measured next to hub plane

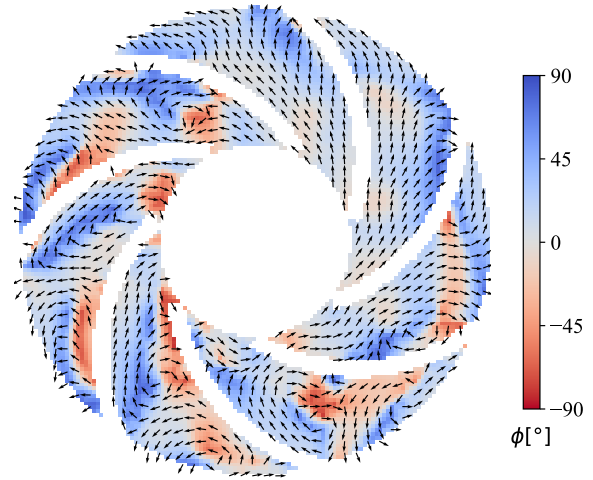


(f) $Q_w = 1.6 Q_{BEP}$ measured next to shroud plane

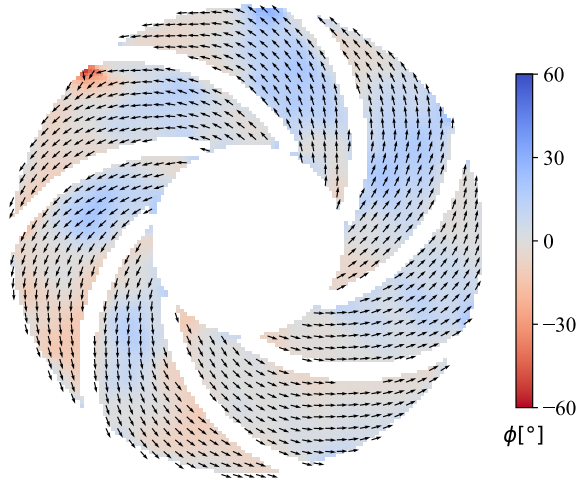
Fig. 9 Angle of deviation ϕ between central and peripheral velocity vectors for $N = 600$ rpm.



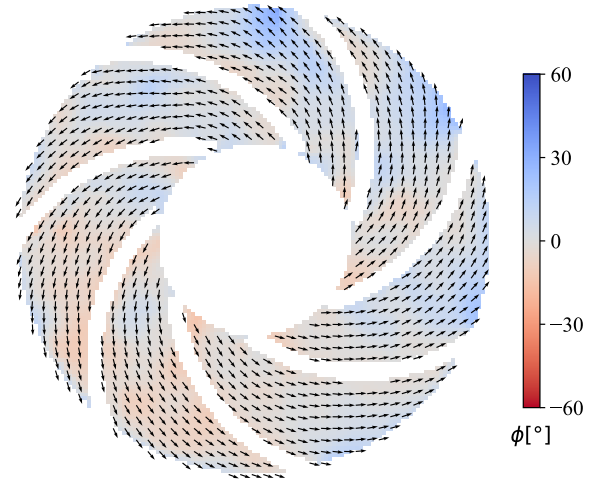
(a) $Q_w = 0.1 Q_{BEP}$ measured next to hub plane



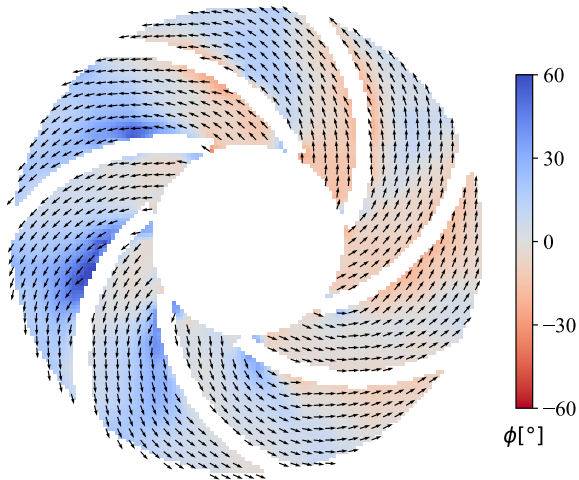
(b) $Q_w = 0.1 Q_{BEP}$ measured next to shroud plane



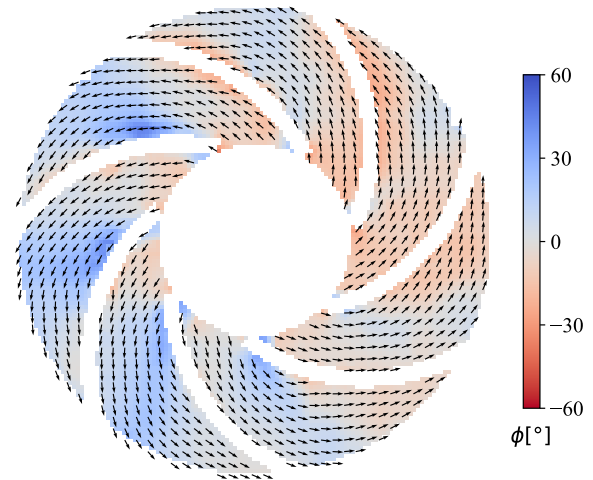
(c) $Q_w = 1.0 Q_{BEP}$ measured next to hub plane



(d) $Q_w = 1.0 Q_{BEP}$ measured next to shroud plane



(e) $Q_w = 1.6 Q_{BEP}$ measured next to hub plane



(f) $Q_w = 1.6 Q_{BEP}$ measured next to shroud plane

Fig. 10 Angle of deviation ϕ between central and peripheral velocity vectors for $N = 900$ rpm.

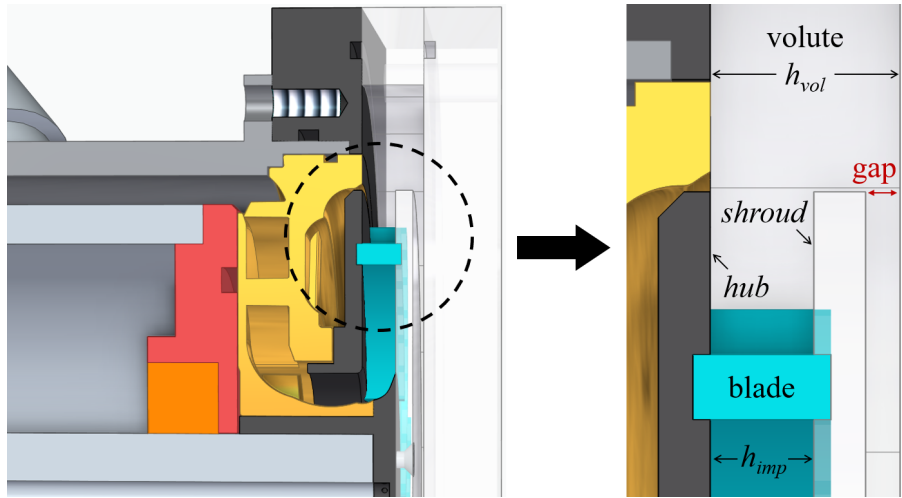


Fig. 11 Technical drawing of the pump stage. Cutaway view highlighting the impeller assembled in the volute.

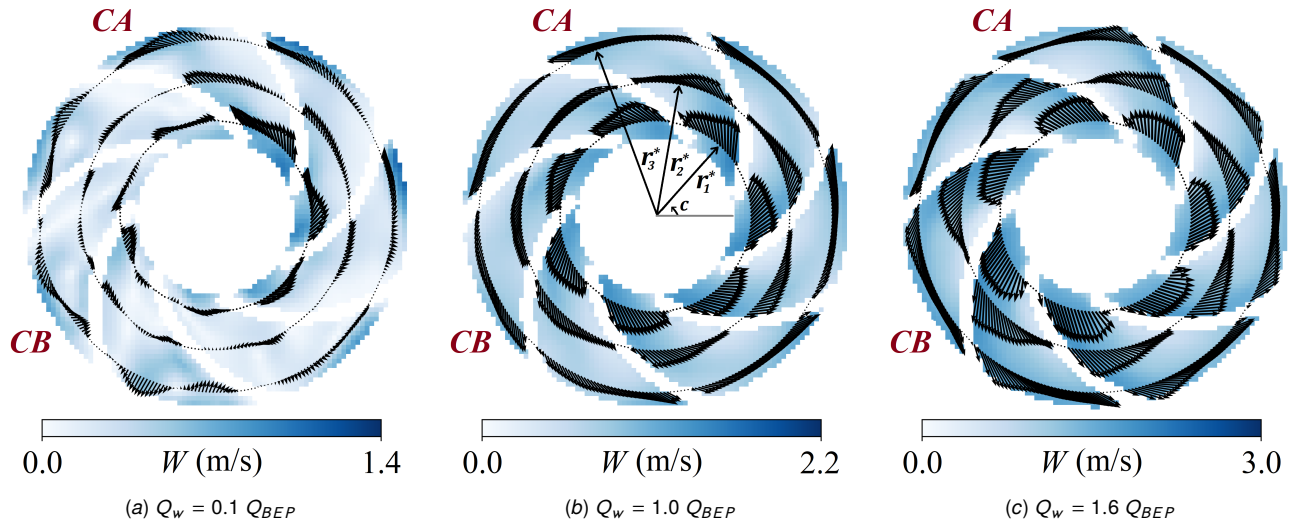
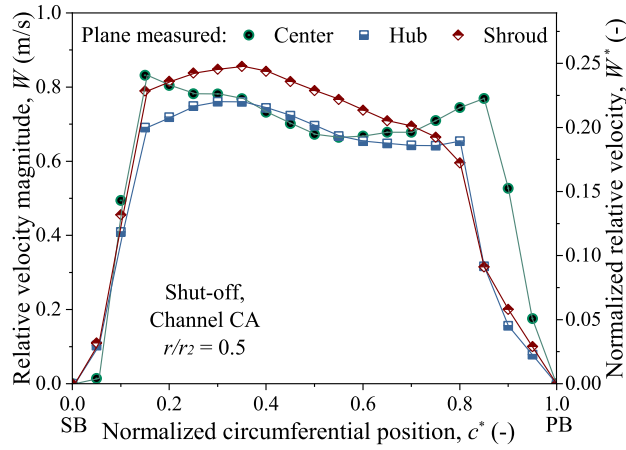
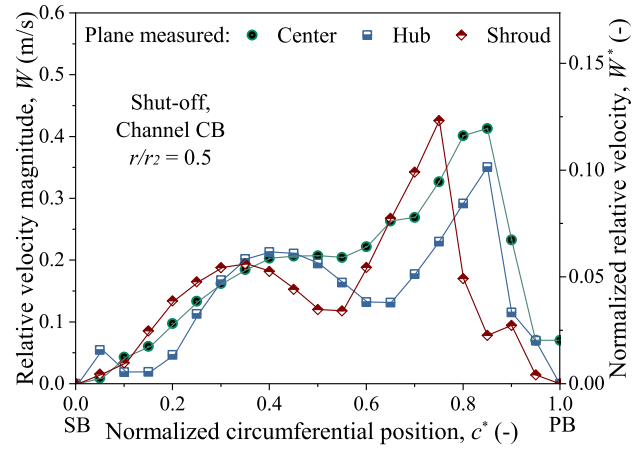


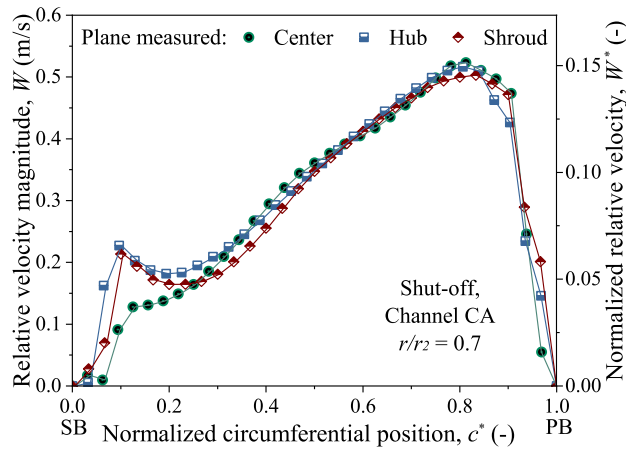
Fig. 12 Velocity vectors W_{center} (black) superimposed on velocity magnitudes W_{center} (blue) in the 3 radial positions $r^* = r/r_2 = [0.5, 0.7, 0.9]$ for $N = 600$ rpm. In this example, measurements refer to the central plane of the impeller.



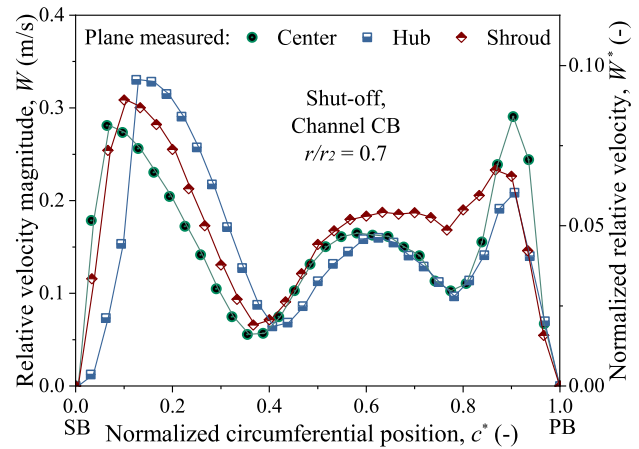
(a) $Q_w = 0.1 Q_{BEP}$, channel CA, $r_1^* = 0.5$



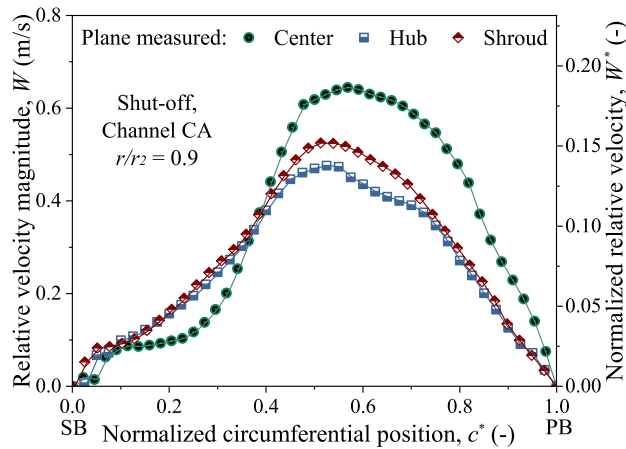
(b) $Q_w = 0.1 Q_{BEP}$, channel CB, $r_1^* = 0.5$



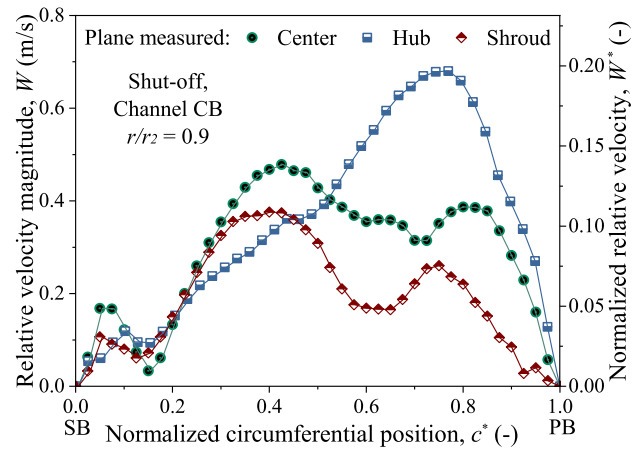
(c) $Q_w = 0.1 Q_{BEP}$, channel CA, $r_2^* = 0.7$



(d) $Q_w = 0.1 Q_{BEP}$, channel CB, $r_2^* = 0.7$

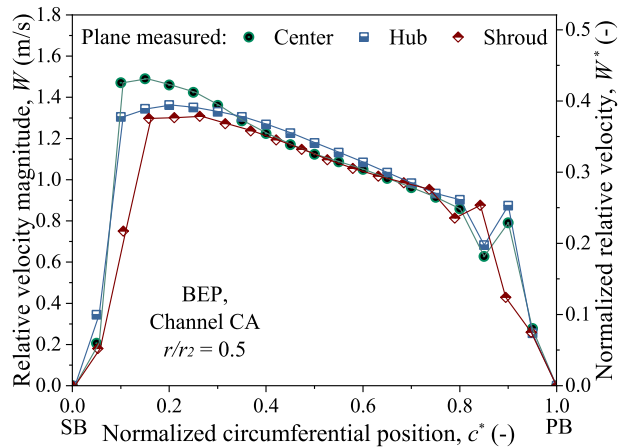


(e) $Q_w = 0.1 Q_{BEP}$, channel CA, $r_3^* = 0.9$

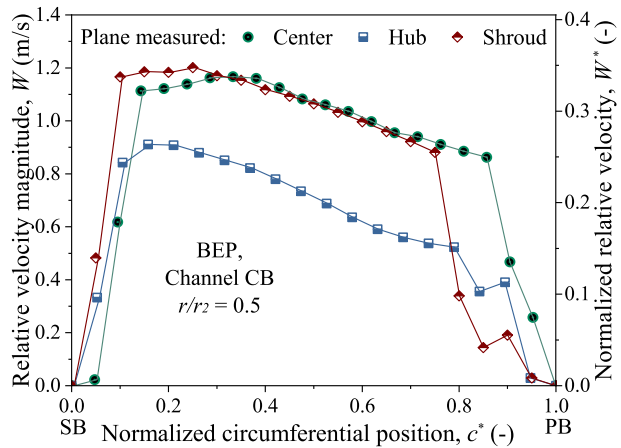


(f) $Q_w = 0.1 Q_{BEP}$, channel CB, $r_3^* = 0.9$

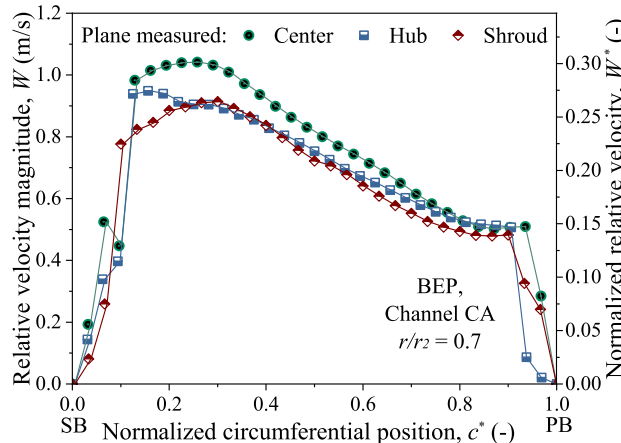
Fig. 13 Profiles of relative velocity, W , and normalized relative velocity, W^* , measured next to central and peripheral planes in the impeller. Shut-off condition at $N = 600$ rpm.



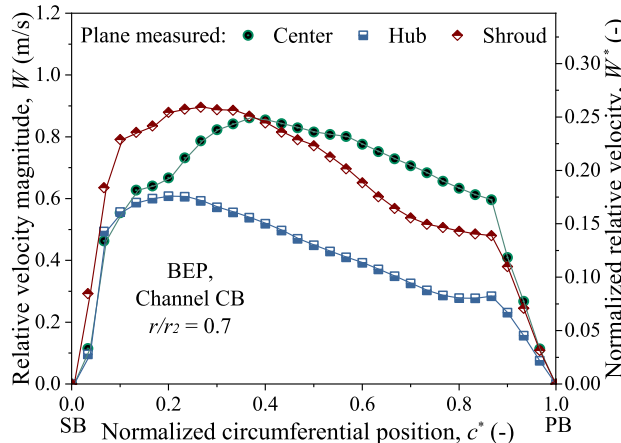
(a) $Q_w = 1.0 Q_{BEP}$, channel CA, $r_1^* = 0.5$



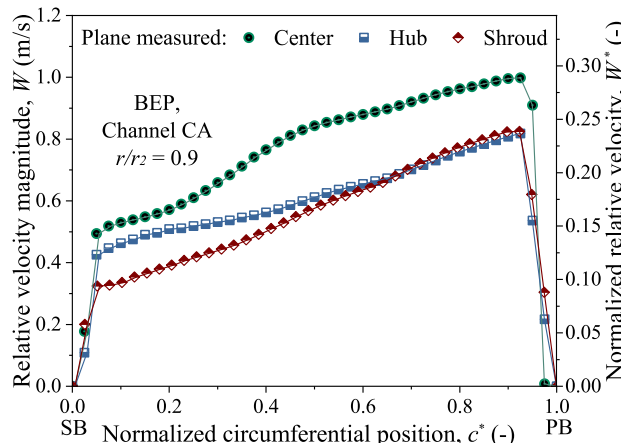
(b) $Q_w = 1.0 Q_{BEP}$, channel CB, $r_1^* = 0.5$



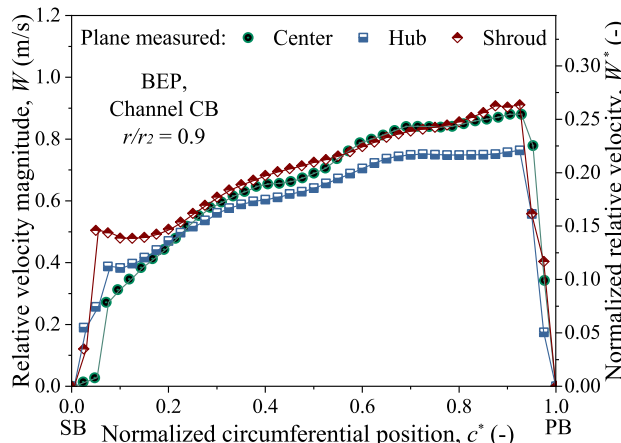
(c) $Q_w = 1.0 Q_{BEP}$, channel CA, $r_2^* = 0.7$



(d) $Q_w = 1.0 Q_{BEP}$, channel CB, $r_2^* = 0.7$

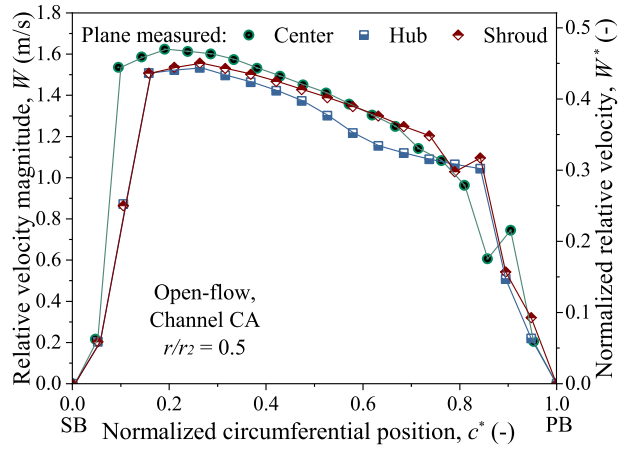


(e) $Q_w = 1.0 Q_{BEP}$, channel CA, $r_3^* = 0.9$

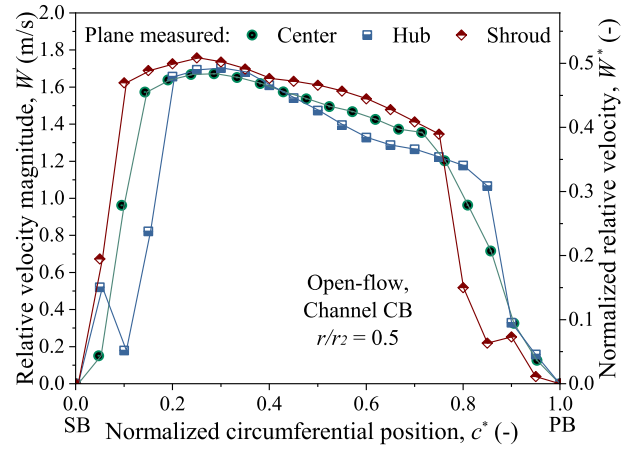


(f) $Q_w = 1.0 Q_{BEP}$, channel CB, $r_3^* = 0.9$

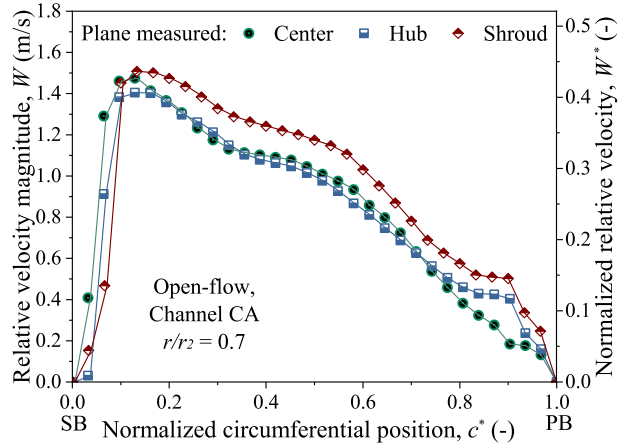
Fig. 14 Profiles of relative velocity, W , and normalized relative velocity, W^* , measured next to central and peripheral planes in the impeller. BEP condition at $N = 600$ rpm.



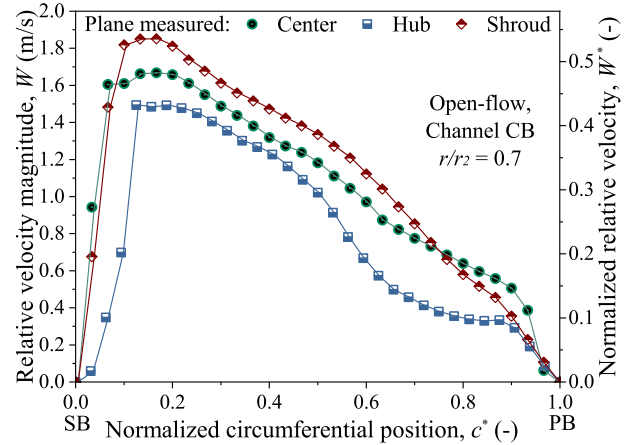
(a) $Q_w = 1.6 Q_{BEP}$, channel CA, $r_1^* = 0.5$



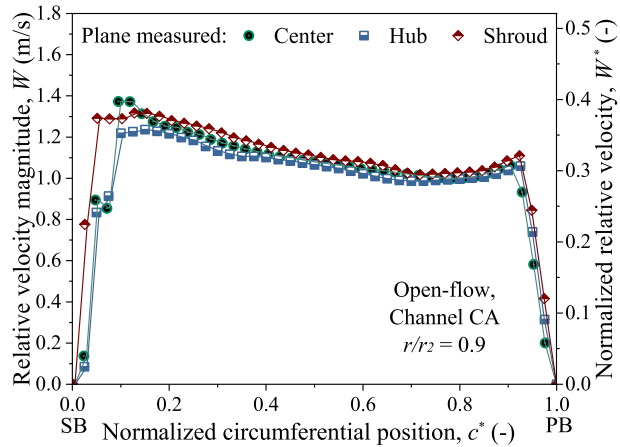
(b) $Q_w = 1.6 Q_{BEP}$, channel CB, $r_1^* = 0.5$



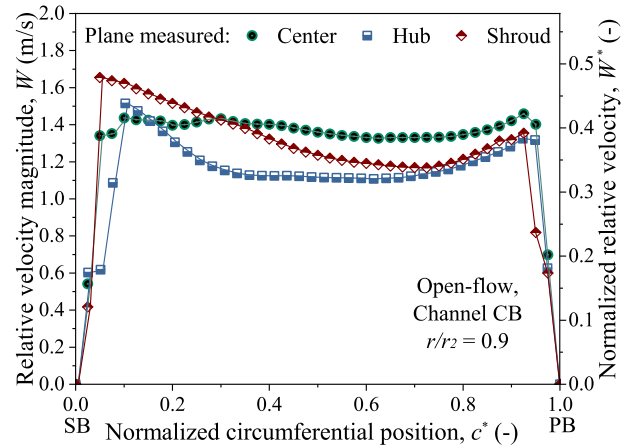
(c) $Q_w = 1.6 Q_{BEP}$, channel CA, $r_2^* = 0.7$



(d) $Q_w = 1.6 Q_{BEP}$, channel CB, $r_2^* = 0.7$



(e) $Q_w = 1.6 Q_{BEP}$, channel CA, $r_3^* = 0.9$



(f) $Q_w = 1.6 Q_{BEP}$, channel CB, $r_3^* = 0.9$

Fig. 15 Profiles of relative velocity, W , and normalized relative velocity, W^* , measured next to central and peripheral planes in the impeller. Open-flow condition at $N = 600$ rpm.

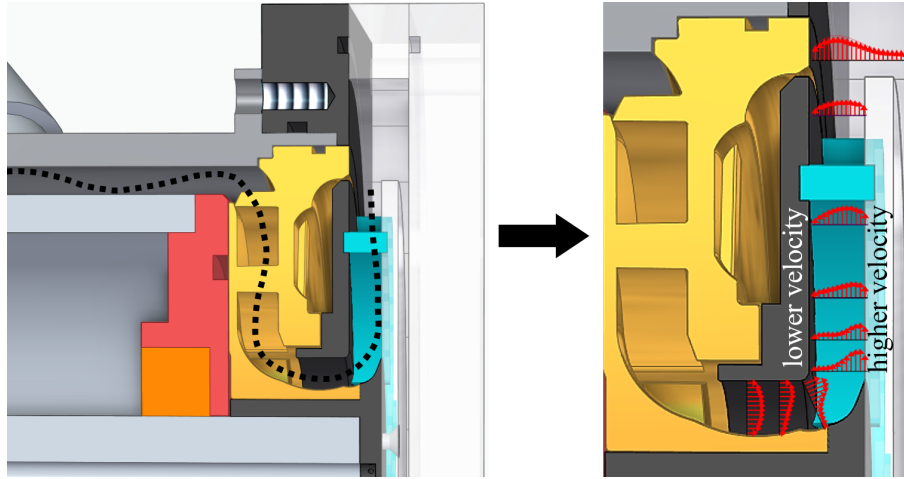
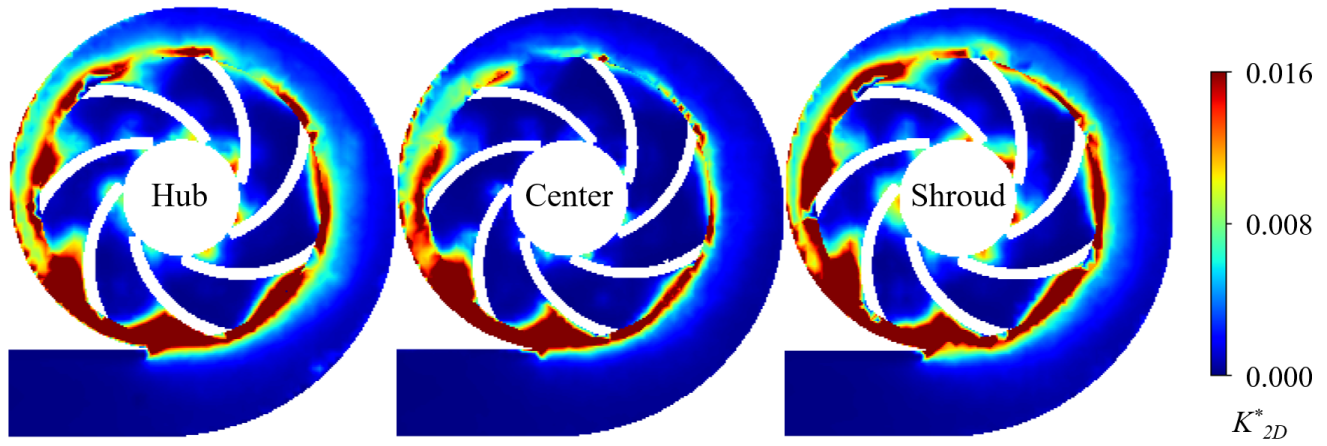
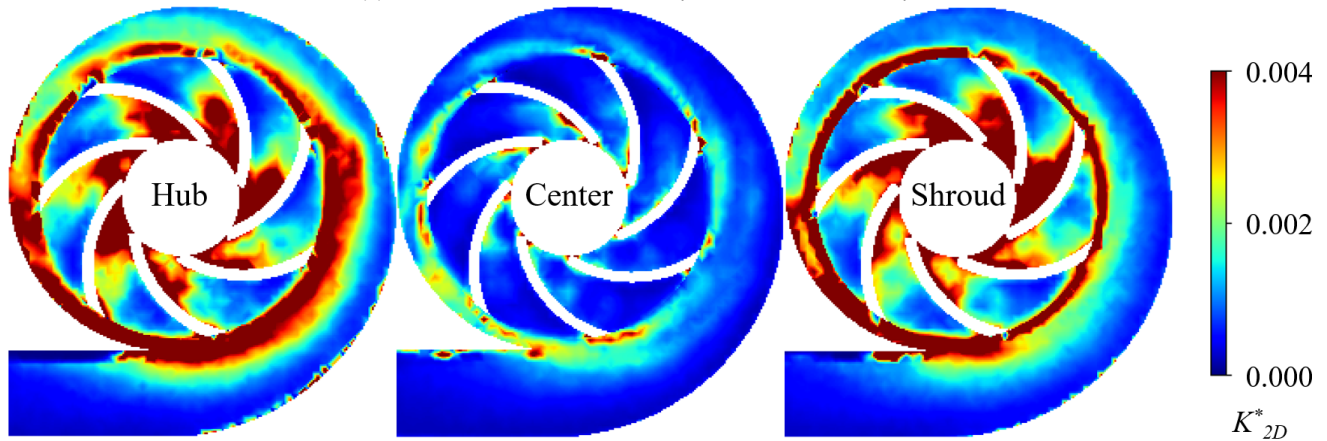


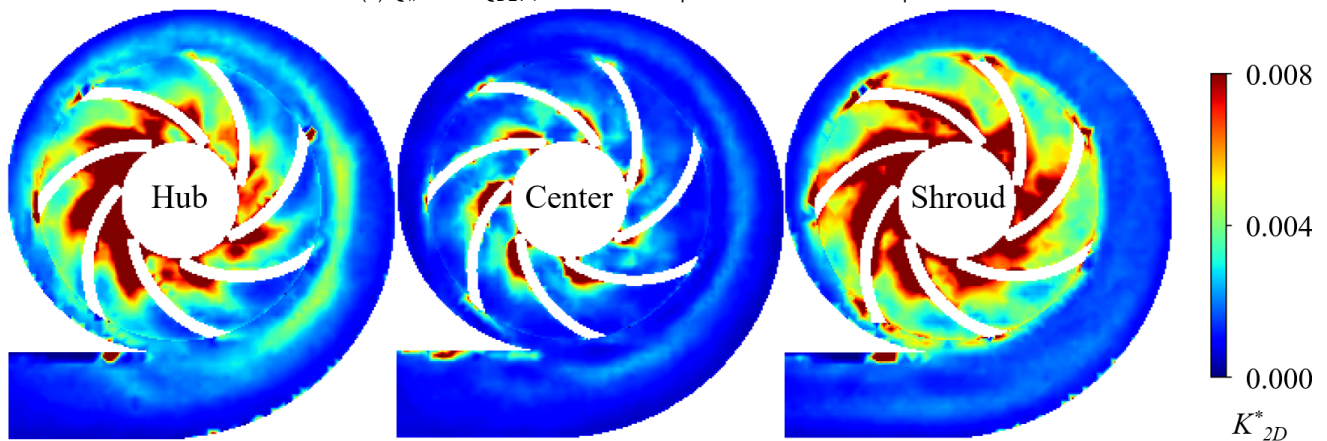
Fig. 16 Cutaway view of the pump. The fluid path (left) undergoes a change in its direction before entering the impeller. This fact causes the velocity profiles (right) to become distorted. Note: the profiles are purely illustrative.



(a) $Q_w = 0.1 Q_{BEP}$, measurements performed on three axial planes



(b) $Q_w = 1.0 Q_{BEP}$, measurements performed on three axial planes



(c) $Q_w = 1.6 Q_{BEP}$, measurements performed on three axial planes

Fig. 17 Contour plots of normalized turbulent kinetic energy, K_{2D}^* , measured on the central and the peripheral planes in the impeller and volute. $N = 600$ rpm.

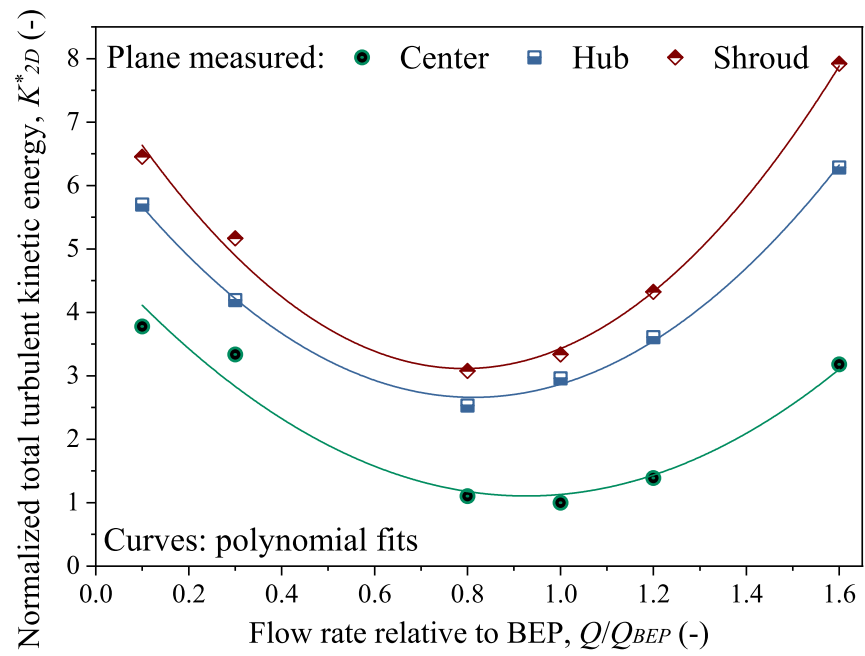


Fig. 18 Total turbulent kinetic energy in the entire impeller, on three planes and six flow rates. $N = 600$ rpm.